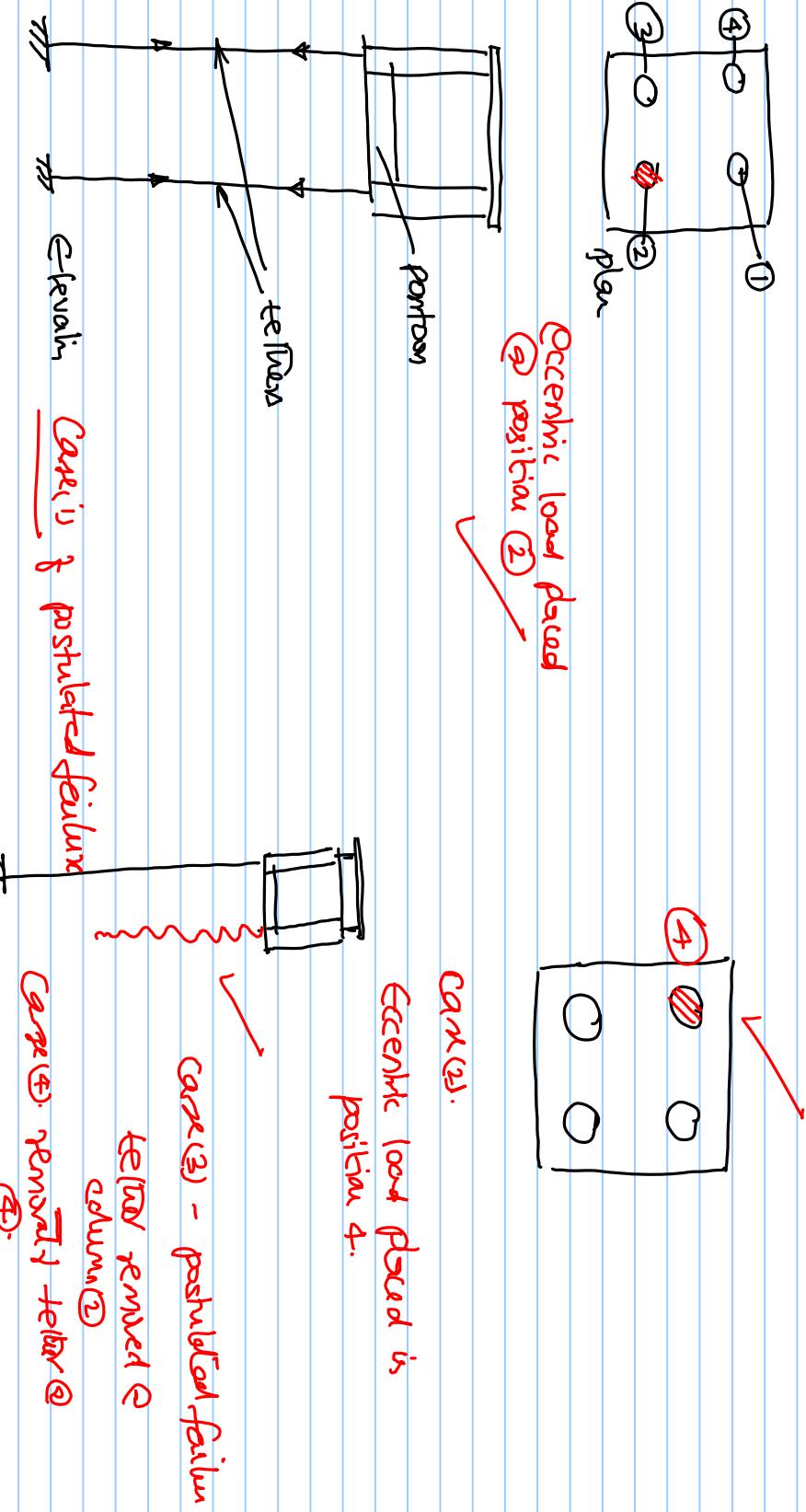


Module 4

Reduce &

✓ Sth or Typ - III
Class scale)



Postulated failure cases are introduced

To examine the 7 AMs

- physical model should not be damaged
 - force amplitude are kept very small.
 - no permanent damage is caused.
- plots depict the eccentric load, which is a common scenario in offshore structures.
- failure cases are referred to - failure } happens in a compliant system like Top- Con Cavitational failure.

Moving system

- load record
- high initial eccentricity
- under axial tension

Sensors

- ④ Sensors are deployed
- post to the sensor is changed for each set of the experiment

Assumption. In each postulated failure case, it is assumed that failure alone occurs in the platform.
(cumulative effect is ignored)

Sensor location

- closer, the more response is measured
- located @ the main center of the disk.

Data processing

- signal-based data analysis
- processing the significant variations of the acquired time history data
- alternatively, a frequency spectrum

Signal-based data analysis
processes can be classified as

- (1) feature extractors
- (2) pattern recognizers

feature extraction process - processing the raw system data to extract sensitive damage features

- In the case of dealing with large data from multiple sensors,
 - these processes condense the data into
 - small set
 - unprocessed w/ statistical tools

frequency domain techniques

- to analyze the stationary evap, which's localized in time domain

- Fast Fourier Transfer (FFT)

- Power Spectral Density (PSD)
- Shorttime Fourier Transfer (STFT)

One want to analyze
the data in
frequency domain

FFT is used to identify the frequency components present in signal.

Let $x(t)$ be time-varying function, which represents the acceleration
time history t → measured (acquired) from the sensor during
experiments

Fourier transform $x(t)$ → analysis

$$X(F) = \int_{-\infty}^{\infty} x(t) e^{-j2\pi F t} dt \quad - (1)$$

- FT decompose the signal into weighted combination of sinusoids + direct frequency
- Transform finds the amplitude & phase difference the sinusoids.

for a specific value $\gamma(F)$

signal is converted with the basis func $e^{-j2\pi ft}$

- values of rays from $-\infty$ to ∞
 - complex constant coeff, known for the value $\gamma(2\pi f)$
- is called Fourier Transform coeff.

- PSD $\gamma(F)$ thus direct represents distribution of power across different frequency present in the signal

$$S_x(f) = \lim_{T \rightarrow \infty} E \left\{ \frac{1}{2T} |X(t)|^2 \right\} = \lim_{T \rightarrow \infty} \left[\frac{1}{2T} \int_{-T}^T x(t) e^{-j2\pi ft} dt \right]^2$$

→ (2)

Eqn) can be interpreted as

expected value of FST of the SNP, computed over an infinite period

- In FT, only global features of the trend are extracted in the frequency axis
- There's no localization of the features across the time axis
 - major deficit in FST
- Transformation is the result - summary statistic across the entire length
- a very frequent - recursive but a poor time - results

In case STM, FFT can identify the damage by measuring frequency spikes

- this damage thus detected is due to breakage in the estimator estimated from the freq value
- information on how contact is lost -

Alternately, STFT

- This actually slices the signal into different segments
- using a window function etc

Each these signs are subjected to P9

$$= x(\tau, t) = \chi(t) w(t-\tau) \quad \text{--- (3)}$$

where t is the window function.

- Window function is pleased sumster

Converg window coincide with short time theory
and it traverses along the curve of the trend.

$$x(\tau, e) = \int x(\tau, \epsilon) \bar{e}^{jet} d\epsilon \quad \text{--- (4)}$$

$$x(\tau, e) = \int x(\epsilon) \bar{w}(t-\tau) \bar{e}^{jet} d\epsilon \quad \text{--- (5)}$$

τ is the central window time, e is the mean frequency width

$\omega(t-\tau)e^{-j\omega t}$ is the SFT - overlaying function.

Windows should have compact support

- It should exit only over a finite time
- vanish outside the interval

- If the window is too long \equiv length of the signal
this process will convert the FFT

Answer: STFT \Rightarrow frequency

$$\begin{aligned} X(\omega) &= \frac{1}{2\pi} \int \int X(\tau, \omega) e^{j\omega\tau} d\tau \\ &= \frac{1}{2\pi} \int \int [x(\tau) w(t-\tau)] e^{j\omega\tau} \omega(t-\tau) e^{-j\omega t} d\tau d\omega \end{aligned} \quad (2)$$

Spektrum - squared magnitude - (P)

- Spektrum ist die Energy density in the time-frequency plane

Energy decomposes in time & frequency:

$$\int |x(\tau)|^2 d\tau = \frac{1}{2\pi} \int |x(\omega)|^2 d\omega$$

$$= \frac{1}{2\pi} \int \int |x(\tau, \omega)|^2 d\omega d\tau = (P)$$

Spektrum, $S(\tau, \omega)$ ist definiert

$$S(\tau, \omega) = |x(\tau, \omega)|^2 = \left| \int x(\tau') e^{-j\omega(\tau - \tau')} d\tau' \right|^2 = (Q)$$

Validate it in developed software

In order to validate the results, we have to measure response of selected trip, is acquired in lots wireless term

- Results are compared, lots as time & frequency domain
to estimate error of distance measurement

- Wind sensor - connected to DAQ through wires

- data is processed @ central server, which is connected to DAQ

- Wireless sensors

- low-cost computing nodes & sensor units

- Acquire data with a transmitter with transmitter estimator

Specification of accelerometer

Wired

Acc 393 B04

Type Infrared Drift sensor
elec.

4 bars

one

Range $\pm 5g$

$\pm 16g$ (oper for $\pm 2g$)

Sensitivity $1V/g$

$16.284 \text{ LSB}/g$

Noise performance $0.3 \text{ mg}/\sqrt{\text{Hz}}$

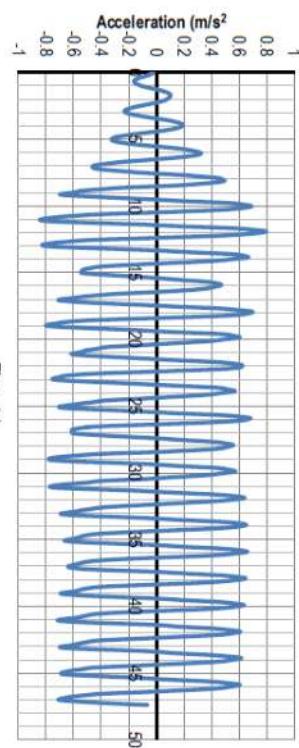
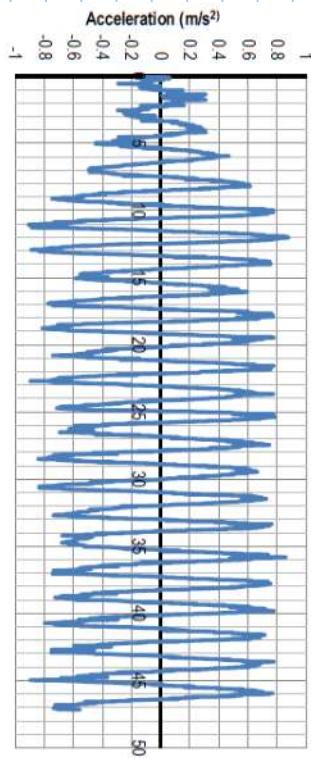
$400 \text{ mg}/\sqrt{\text{Hz}}$

Wireless

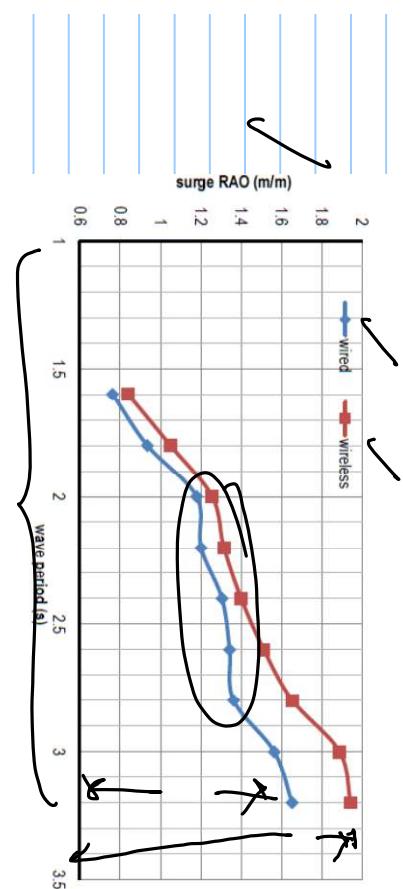
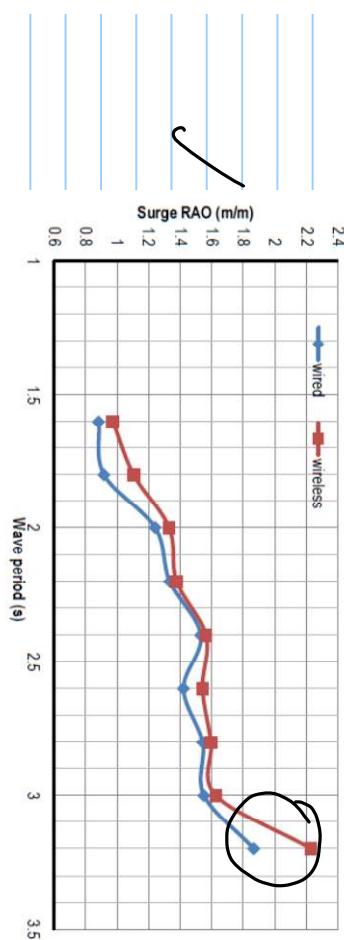
MIMO 6050

MEMS

3



a) Wired acquisition



b) Wireless acquisition



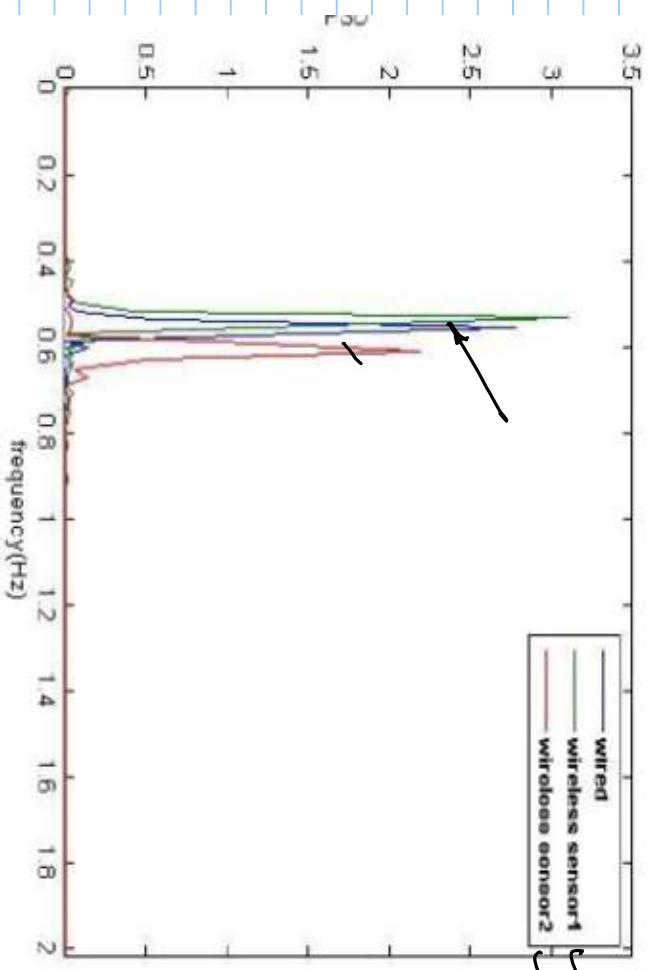
a) surge response.

— On comparison the wired wireless % error is higher for higher periods.



— wired/wireless variation is less -

Wave Period (S)	WH = 8cms			WH = 10cms			WH = 12cms		
	Wired	Wireless	Diff in %	Wired	Wireless	Diff in %	Wired	Wireless	Diff in %
1.6	0.76	0.84	10.00	0.89	0.98	10.18	0.89	0.97	9.71
1.8	0.94	1.05	12.28	0.92	1.01	9.58	0.99	1.13	13.67
2	1.12	1.24	6.53	1.25	1.33	7.15	1.18	1.28	8.59
2.2	1.20	1.31	9.43	1.34	1.38	3.26	1.31	1.44	9.36
2.4	1.31	1.40	7.33	1.54	1.57	2.17	1.34	1.53	14.59
2.6	1.34	1.51	12.62	1.42	1.55	8.47	1.42	1.55	9.73
2.8	1.36	1.66	21.41	1.54	1.59	3.44	1.34	1.73	29.77
3	1.57	1.89	20.70	1.55	1.63	4.87	1.53	1.98	29.82
3.2	1.66	1.94	17.57	1.87	2.22	19.50	1.58	1.96	23.46



- PSD - after part - problems in data
- There is a marginal variance in terms of global features/interior to the location
- peak frequency acquired is lots wired & wireless (STM-II)
- marginal difference

- STM-II WSN
- Variance with wired - 10%.
- Shift in frequency - due to time lag
or, the response of the system

PSD of Surge response.

Reliability test results

- Reliability problem is formulated
- assumptions

- 1) peak amplitude of the acquired response under normal conditions (no perturbed faulty)
 - at threshold value $\underline{\underline{=}}$

- 2) if response amplitude, acquired during the perturbed failure case exceeds this value, we need to activate the alert limits sys (AHS)

System failure is defined as user-induced perturbed failure. (Progressive failure is not considered)

part - process the data.
skewness ↗

Summary

- experimental invert & typ (classical)

wired
unwired

