







## Module 3: Internal Flows

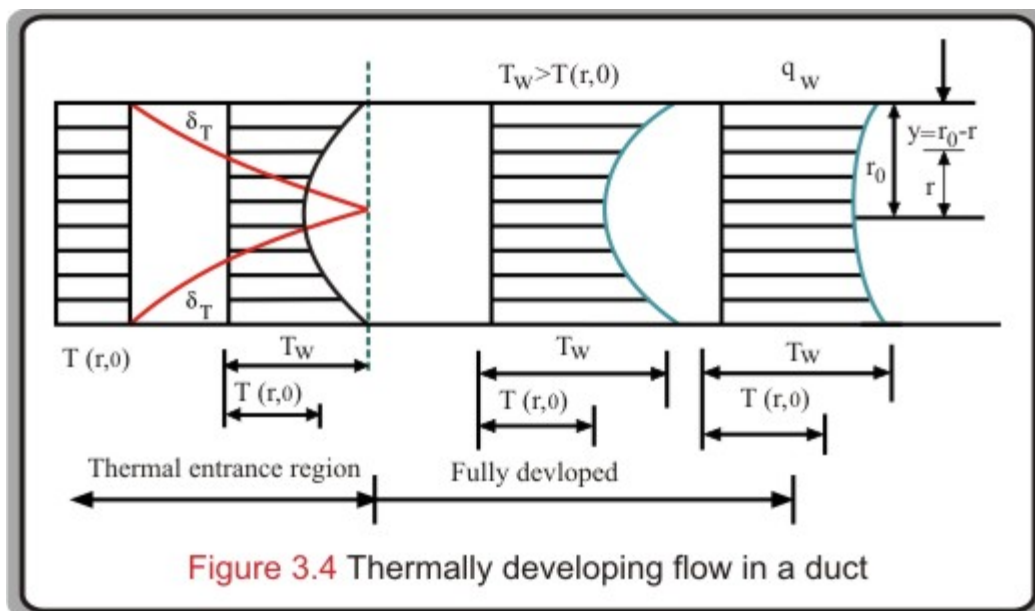
### Lecture 14: Physical Explanation of Surface Conditions

#### The Lecture Contains:

-  Thermal Considerations during Internal Flows
-  Newton's law of cooling
-  Thermal fully developed conditions
-  Energy balance in ducted flows
-  Constant surface temperature
-  Mathematical relation for variation for constant wall temperature

 **Previous**   **Next** 

## Thermal Considerations during Internal Flows



Tube surface condition: either a uniform wall temperature (**UWT**) condition or a uniform wall heat flux condition (**UWH**). A fully developed temperature profile differs according to the surface conditions. For

both surface conditions, however, the amount by which the fluid temperature exceeds the entrance temperature increases with increasing  $(x)$  (**Figure 3.4**). In order to study the development of thermal boundary layer, a hydrodynamically fully developed flow may be considered. (**figure 3.5**)

➤ 1. For laminar flow, the thermal entrance length may be expressed as

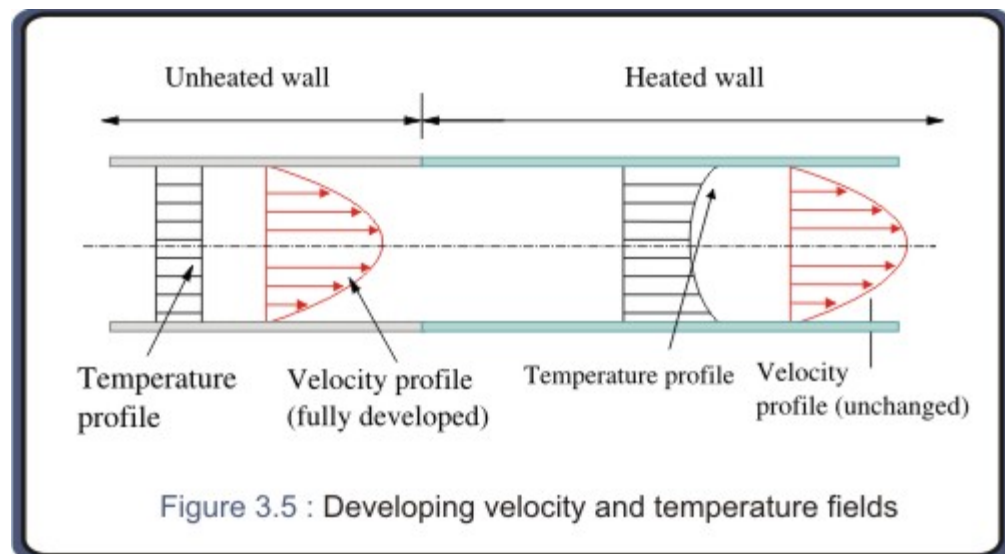
$$\left( \frac{x_{e,t}}{D} \right)_{\text{lam}} \approx \begin{cases} 0.033 Re_D Pr & \text{for UWH} \\ 0.043 Re_D Pr & \text{for UWT} \end{cases} \quad (3.20)$$

we can say,  $(x_{e,t}/D) = 0.05 Re_D Pr$

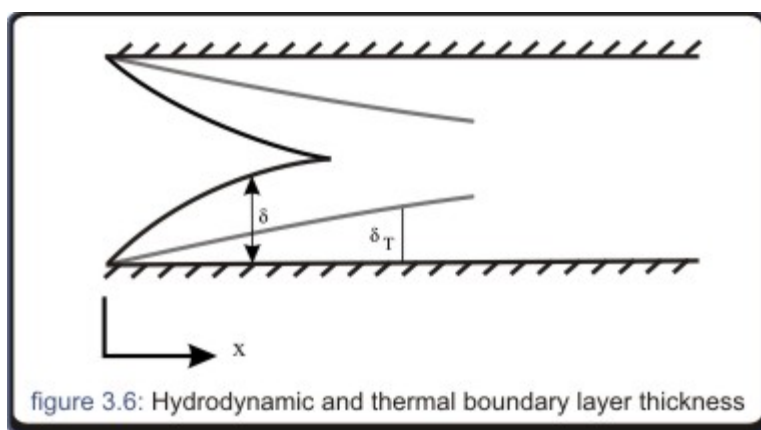
Comparing this with hydrodynamic boundary layer, it can be said that if  $Pr > 1$ , hydrodynamic boundary layer grows more rapidly.

$Pr > 1$ ,  $x_{e,h} < x_{e,t} \Rightarrow \delta > \delta_T$  at any section; even otherwise  $Pr \sim (\delta/\delta_T)^{1/2}$ .

Therefore, for  $Pr > 1$ ,  $\delta$  has to be  $> \delta_T$ .



Inverse is true for  $Pr < 1, Pr \sim (\delta/\delta_T)^{1/3}$  and consequently  $\delta_T$  is greater than  $\delta$ . However, for extremely High Prandtl number fluids (such as oil),  $Pr \geq 100$ ,  $x_{e,h} \ll x_{e,t}$ . Throughout the thermal entrance region, hydrodynamically fully developed velocity profile can be assumed (figure 3.6).



**figure 3.6:** Hydrodynamic and thermal boundary layer thickness  $Pr \gg 1$  For turbulent flow, conditions are nearly independent of Prandtl number. It is fair enough to accept  $(\frac{x_{e,t}}{D})_{tur} = 10$ . Bulk mean temperature at any section ( $x$ ) is given by

$$T_m = \frac{2}{u_m r_0^2} \int_0^{r_0} u(r) T(r) r dr \quad (3.21)$$

◀ Previous Next ▶

**Newton's law of cooling**

$$q_w'' = h(T_w - T_m) \quad (3.22)$$

The bulk mean temperature  $T_m$  is a convenient reference temperature for internal flows.  $T_m$  behaves like  $T_\infty$  for external flows. Essential difference is  $T_m$  varies in the flows direction.  $T_m$  is increased in the flows direction if heat transfer is from surface to fluid.

**Thermally Fully developed conditions**

Since the existence of convective heat transfer between the surface and the fluid dictates that the fluid temperature must continue to change with  $x$ , one might question whether fully developed thermal conditions ever can be reached. The situation is certainly different from the hydrodynamic case, for which  $(\partial u / \partial x) = 0$  in the fully developed region. **In contrast, if there is heat transfer,  $(dT_m / dx)$  as well as  $(\partial T / \partial x)$  at any radius is not zero.**

**Introducing a dimensionless temperature**  $(T_w - T) / (T_w - T_m)$ , condition for which this ratio becomes independent of  $x$  are known to exist. Although the temperature profile  $T(r)$  continues to change with  $x$ , the relative shape of the profile does not change and the flow is said to be fully developed. Instead of  $(dT_m / dx) = 0$  or  $(dT / dx) = 0$ , the condition is

$$\frac{d}{dx} \left[ \frac{T_w(x) - T(r, x)}{T_w(x) - T_m(x)} \right] = 0 \quad (3.23)$$

i.e.,  $(T_w - T)$  changes in the same way as  $(T_w - T_m)$

◀ Previous   Next ▶

## Module 3: Internal Flows

## Lecture 14: Physical Explanation of Surface Conditions

Now, we can write

$$\frac{(T_w - T_m) \left( \frac{dT_w}{dx} - \frac{dT}{dx} \right) - (T_w - T) \left( \frac{dT_w}{dx} - \frac{dT_m}{dx} \right)}{(T_w - T_m)^2} = 0$$

or

$$\frac{1}{(T_w - T_m)} \frac{dT_w}{dx} - \frac{dT}{dx} \frac{1}{(T_w - T_m)} - \frac{T_w - T}{(T_w - T_m)^2} \frac{dT_w}{dx} + \frac{(T_w - T)}{(T_w - T_m)^2} \frac{dT_m}{dx} = 0$$

or

$$\frac{dT_w}{dx} - \frac{dT}{dx} - \frac{T_w - T}{T_w - T_m} \frac{dT_w}{dx} + \frac{(T_w - T)}{(T_w - T_m)} \frac{dT_m}{dx} = 0$$

or

$$\frac{dT}{dx} = \frac{dT_w}{dx} - \frac{T_w - T}{T_w - T_m} \frac{dT_w}{dx} + \frac{(T_w - T)}{(T_w - T_m)} \frac{dT_m}{dx} \quad (3.24)$$

Two surface conditions are available. **uniform heat flux**, ( $q_w'' = \text{constant}$ ) and **uniform surface temperature**, ( $dT_w/dx = 0$ ).

For ( $q_w'' = \text{constant}$ ): If the tube wall were heated electrically or, if the outer surface were uniformly irradiated.

For ( $dT_w/dx = 0$ ): if the tube were occurring at the outer surface.

However, **for constant heat flux**, we get

$$\frac{dT_w}{dx} = \frac{dT_m}{dx}$$

◀ Previous   Next ▶

## Module 3: Internal Flows

## Lecture 14: Physical Explanation of Surface Conditions

Substituting this condition in the above equation for  $dT/dx$ , we get

$$\left. \frac{dT}{dx} \right|_{\text{fully developed}} = \left. \frac{dT_m}{dx} \right|_{\text{fully developed}} \quad (3.26)$$

For constant surface temperature,  $\frac{dT_w}{dx} = 0$

$$\left. \frac{dT}{dx} \right|_{f,d,t} = \frac{(T_w - T)}{(T_w - T_m)} \left. \frac{dT_m}{dx} \right|_{f,d,t} \quad (3.27)$$

Several important features of thermally developed flow may be inferred from **equation (3.24)**. Since nondimensional temperature profile is independent of  $x$ , derivative of the profile with respect to  $r$  (gradient) must also be independent of  $x$ . **Evaluating this derivative at the tube surface, we obtain**

$$\frac{\partial}{\partial r} \left[ \frac{T_w - T}{T_w - T_m} \right]_{r=r_0} = \frac{-\partial T / \partial r|_{r=r_0}}{(T_w - T_m)} \neq f(x) \quad (3.28)$$

Again we know

$$q_w'' = k \frac{\partial T}{\partial r} \Big|_{r=r_0} = h(T_w - T_m)$$

or,

$$\frac{-\frac{\partial T}{\partial r} \Big|_{r=r_0}}{(T_w - T_m)} = \frac{h}{k} \quad (3.29)$$

From (3.28) and (3.29), we get,  $\frac{h}{k} \neq f(x)$ . for fully developed flow (**Figure 3.7**). Here  $k$  is assumed to be constant, **hence  $h$  is independent of.**

◀ Previous   Next ▶

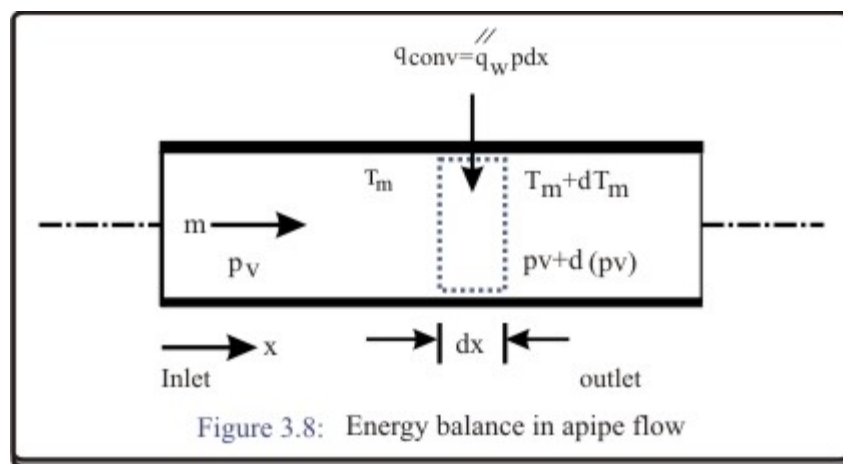
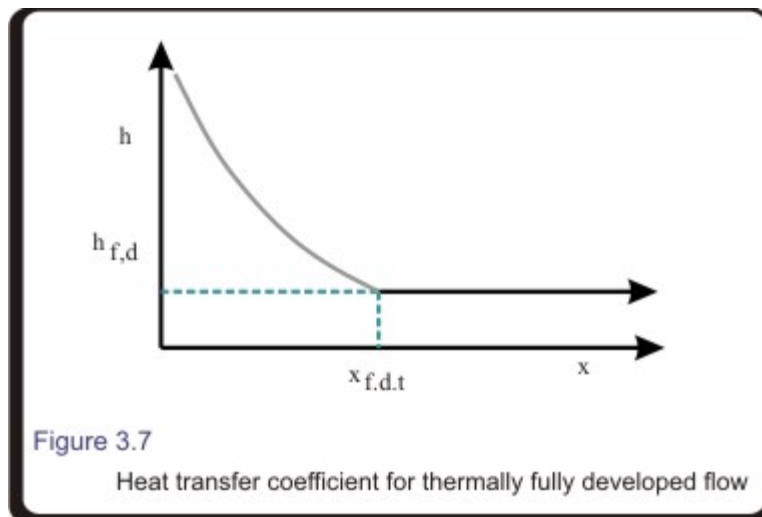
## Module 3: Internal Flows:

## Lecture 14: Physical Explanation of Surface Conditions

## Energy Balance in Ducted Flows

Because the flow in a tube is completely enclosed, an energy balance may be applied to determine how the mean temperature ( $T_m$ ) varies with position along the tube and how the total convective heat transfer is related to the difference in temperatures at the tube inlet outlet. **Consider Figure 3.8.**

$$dq_{conv} + \dot{m}(c_v T_m + pv) - \left[ \dot{m}(c_v T_m + pv) + \dot{m} \frac{d(c_v T_m + pv)}{dx} dx \right] = 0$$



or

$$dq_{conv} = \dot{m} d(c_v T_m + pv) \quad (3.30)$$

For compressible flow:

$$dq_{conv} = \dot{m} d(c_v T_m + pv)$$

or

$$dq_{conv} = \dot{m} d(c_v T_m + R T_m)$$

◀◀ Previous   Next ▶▶



## Module 3: Internal Flows

## Lecture 14: Physical Explanation of Surface Conditions

or,

$$dq_{conv} = \dot{m} c_p dT_m \text{ [since } d(pv) = 0 \text{ and } c_p = c_v]$$

or,

$$dq_{conv} = \dot{m} c_p dT_m \quad (3.31)$$

For incompressible flow:

$$dq_{conv} = \dot{m} c_p dT_m \text{ [since } d(pv) = 0 \text{ and } c_p = c_v]$$

Therefore, **whether the flow is compressible or incompressible**, we can write

$$q_{conv} = \dot{m} c_p (T_{m,0} - T_{m,i}) \quad (3.32)$$

we can also write

$$q_{conv} = \dot{m} c_p (T_{m,0} - T_{m,i})$$

or,

$$\frac{q_w'' P dx}{\pi D} = \dot{m} c_p dT_m \quad [P \text{ is the surface perimeter} = \pi D]$$

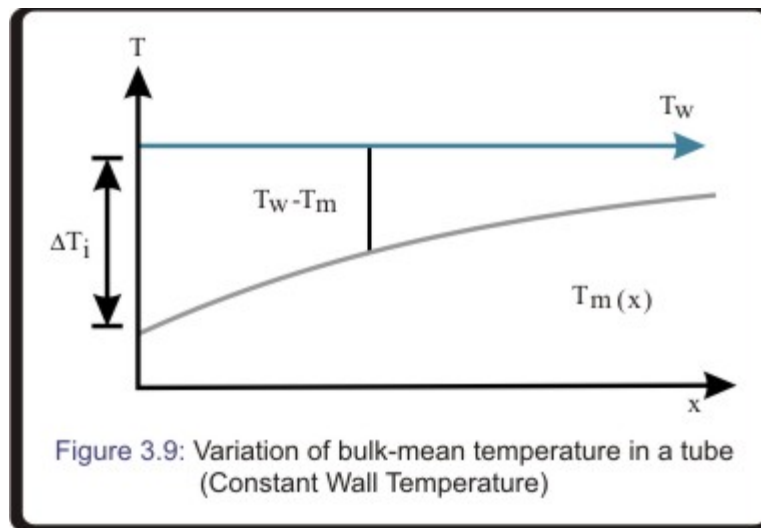
or,

$$\frac{dT_m}{dx} = \frac{q_w'' P}{\dot{m} c_p} = \frac{Ph(T_w - T_m)}{\dot{m} c_p} \quad (3.33)$$

So long,  $T_w > T_m$ , heat is transferred to fluid and  $T_m$  increases with  $x$ . The manner in which **equation (3.33)** varies should be noted.

**Constant surface temperature:**

$P$  may vary with  $x$ , but it is constant for pipe.  $P/\dot{m}c_p$  is constant. For fully developed flow,  $h$  is constant, although it varies with  $x$ , in the entrance region. Finally  $T_w$  is constant. So,  $T_m$  will vary with  $x$  except for the trivial case ( $T_w = T_m$ ) of no heat transfer (Figure 3.9).

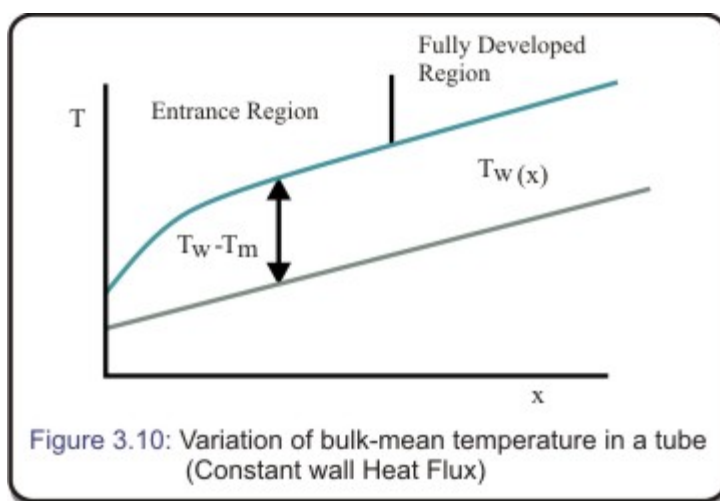
**Constant heat flux:**

$q_w''$  is independent of  $x$ . Total heat transfer rate  $q_{conv}^{total} = q_w'' PL$ ; again, from equation

◀ Previous    Next ▶

## Module 3: Internal Flows

## Lecture 14: Physical Explanation of Surface Conditions



(3.30-3.32) ,  $q_w'' PL = \dot{m} c_p (T_{m,0} - T_{m,i})$  . so if  $q_w''$ ,  $\dot{m}$ ,  $c_p$  and the geometry are known, it is possible to calculate the fluid temperature rise  $(T_{m,0} - T_{m,i})$

otherwise

$$\frac{dT_m}{dx} = \frac{q_w'' P}{\dot{m} c_p} \text{ or } T_m = \frac{q_w'' P}{\dot{m} c_p} x + c_1$$

at  $x = 0$ ,  $T_m = T_{m,i}$  therefore  $c_1 = T_{m,i}$

$$T_m(x) = T_{m,i} + \frac{q_w'' P}{\dot{m} c_p} x \quad (3.34)$$

We can see the variation  $T_m(x)$  with  $x$ , for constant wall heat flux (**Figure 3.10**) .Let us now find out the mathematical relation for variation of  $T_m(x)$  with  $x$ , for constant wall temperature case.

◀ Previous Next ▶

**Mathematical relation for variation of  $T_m(x)$  for constant wall temperature**

$P$  may vary with  $x$ , but it is constant for pipe.  $P/\dot{m}c_p$  is constant. For fully developed flow,  $h$  is constant, although it varies with  $x$ , in the entrance region. Finally  $T_w$  is constant. So,  $T_m$  will vary with  $x$ , except for the trivial case ( $T_w = T_m$ ) of no heat transfer. From equation (3.33),  $\frac{dT_m}{dx} = \frac{q_w'' P}{\dot{m}c_p} = \frac{Ph(T_w - T_m)}{\dot{m}c_p}$ , defining  $\Delta T = T_w - T_m =$  difference between surface temperature and mean temperature.

$$\frac{dT_m}{dx} = -\frac{d(\Delta T)}{dx} = \frac{Ph(\Delta T)}{\dot{m}c_p} \quad (3.35)$$

$$\int_{\Delta T_i}^{\Delta T_0} \frac{d(\Delta T)}{(\Delta T)} = \frac{-P}{\dot{m}c_p} \int_0^L h dx$$

or,

$$\frac{(\Delta T_0)}{(\Delta T_i)} = -\frac{PL}{\dot{m}c_p} \cdot \frac{1}{L} \int_0^L h dx$$

or,

$$\frac{\Delta T_0}{\Delta T_i} = -\frac{PL}{\dot{m}c_p} \bar{h}_L$$

$$\frac{\Delta T_0}{\Delta T_i} = \frac{T_w - T_{m,0}}{T_w - T_{m,i}} = \exp\left(-\frac{PL}{\dot{m}c_p} \bar{h}_L\right) \quad (3.36)$$

◀◀ Previous    Next ▶▶

## Module 3: Internal Flows

## Lecture 14: Physical Explanation of Surface Conditions

Had we integrated from the tube inlet to some axial position  $x$ , within the tube, we would have obtained the similar, but more general expression.

$$\frac{T_w - T_m(x)}{T_w - T_{m,i}} = \frac{T_w - T_m(x)}{T_w - T_{m,i}} = \exp\left(-\frac{Px}{\dot{m}c_p}\bar{h}\right) \quad (3.37)$$

where  $\bar{h}$  is now the average value of  $h$  from the inlet to the distance  $x$ . This result suggests that the temperature difference ( $T_w - T_m$ ) decays exponentially (also clear from the figure shown earlier). we can also write

$$q_{conv} = \dot{m} c_p (T_{m,0} - T_{m,i})$$

or,

$$q_{conv} = \dot{m} c_p [(T_w - T_{m,i}) - (T_w - T_{m,0})]$$

or,

$$q_{conv} = \dot{m} c_p [(T_w - T_{m,i}) - (T_w - T_{m,0})] \quad (3.38)$$

On the other hand we have seen that

$$\ln \frac{\Delta T_0}{\Delta T_i} = -\frac{PL}{\dot{m}c_p} \bar{h}_L \quad (3.39)$$

Substituting for  $\dot{m}c_p$  in (3.38) from (3.39) we get

$$q_{conv} = \frac{\bar{h}_L PL (\Delta T_0 - \Delta T_i)}{\ln(\Delta T_0 / \Delta T_i)} = \bar{h}_{av} A_w \Delta T_{l,m} \quad (3.40)$$

This is a form of **Newton's law of cooling for the entire tube**.  $\Delta T_{l,m}$  is the appropriate average of temperature difference for the entire tube. The log nature of this average temperature difference [in contrast to an arithmetic mean temperature difference of the form  $\Delta T_m = \frac{\Delta T_i + \Delta T_0}{2}$  is due to exponential nature of the curve.

◀ Previous   Next ▶