

# Linear Dynamical Systems

## Tutorial on State Feedback, Part-II

- ① Cyclic Design (Lecture slides 43 – 52)
- ② Cyclic Design (Lecture slides 43 – 52)
- ③ State Feedback Design for Multi-Input system
- ④ State Feedback and Disturbance Rejection (Lecture Slides 27 – 36)
- ⑤ Feedback Invariant of Nonlinear system
- ⑥ Linear Quadratic Regulator(LQR) (Lecture slides 53 – 62)

## Problem 1

Given a plant defined by  $(A, B)$  pair,

$$A = \begin{bmatrix} -1 & 1 & 0 \\ 0 & -1 & 0 \\ 0 & 0 & -1 \end{bmatrix} \quad B = \begin{bmatrix} 2 & 1 \\ 0 & 2 \\ 1 & 0 \end{bmatrix}$$

Is matrix  $A$  cyclic? Can a state feedback controller be designed using single input variable controller design method?

# Solution to Problem 1

## Recall (Lecture Slide 43)

A matrix  $A$  is called *cyclic* whenever the Jordan form of  $A$  has one and only Jordan block associated with each distinct eigenvalue.

For

$$A = \begin{bmatrix} -1 & 1 & 0 \\ 0 & -1 & 0 \\ 0 & 0 & -1 \end{bmatrix}$$

The eigenvalues of  $A$  are  $\{-1, -1, -1\}$ , forming two Jordan blocks of size  $(2 \times 2)$  and  $(1 \times 1)$ . Therefore,  $A$  is not cyclic.

# Solution to Problem 1

## Recall (Lecture Slide 43)

A matrix  $A$  is called *cyclic* whenever the Jordan form of  $A$  has one and only Jordan block associated with each distinct eigenvalue.

For

$$A = \begin{bmatrix} -1 & 1 & 0 \\ 0 & -1 & 0 \\ 0 & 0 & -1 \end{bmatrix}$$

The eigenvalues of  $A$  are  $\{-1, -1, -1\}$ , forming two Jordan blocks of size  $(2 \times 2)$  and  $(1 \times 1)$ . Therefore,  $A$  is not cyclic.

## Recall(Lecture slide 48)

If  $(A, B)$  is controllable, then for almost any  $p \times n$  real constant matrix  $K$ , the matrix  $(A - BK)$  has only distinct eigenvalues and is, consequently cyclic.

Further, we can also calculate the controllability matrix of the pair  $(A, B)$  as

$$\mathfrak{C} = \begin{bmatrix} 2 & 1 & -2 & 1 & 2 & -3 \\ 0 & 2 & 0 & -2 & 0 & 2 \\ 1 & 0 & -1 & 0 & 1 & 0 \end{bmatrix}$$

which has a rank equal to 3. Therefore, pair  $(A, B)$  is controllable.

# Solution to Problem 1

Recall(Lecture slide 48)

If  $(A, B)$  is controllable, then for almost any  $p \times n$  real constant matrix  $K$ , the matrix  $(A - BK)$  has only distinct eigenvalues and is, consequently cyclic.

Suppose  $K$  is arbitrarily selected as

$$K = \begin{bmatrix} 1 & 0.5 & -1 \\ -1 & 0.8 & 1 \end{bmatrix}$$

Then,

$$\begin{aligned} (A - BK) &= \begin{bmatrix} -1 & 1 & 0 \\ 0 & -1 & 0 \\ 0 & 0 & -1 \end{bmatrix} - \begin{bmatrix} 2 & 1 \\ 0 & 2 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} 1 & 0.5 & -1 \\ -1 & 0.8 & 1 \end{bmatrix} \\ &= \begin{bmatrix} -2 & -0.8 & 1 \\ 2 & -2.6 & -2 \\ -1 & -0.5 & 0 \end{bmatrix} \end{aligned}$$

which can be written in the Jordan canonical form as  $\begin{bmatrix} -1.6 & 0 & 0 \\ 0 & -2 & 0 \\ 0 & 0 & -1 \end{bmatrix}$ .

Therefore,  $(A - BK)$  is cyclic.

## Problem 2

Given a plant defined by  $(A, B)$  pair,

$$A = \begin{bmatrix} -1 & 1 & 0 \\ 3 & 2 & 0 \\ 2 & 0 & 4 \end{bmatrix} \quad B = \begin{bmatrix} 2 & 1 \\ 0 & 2 \\ 1 & 0 \end{bmatrix}$$

Is  $A$  cyclic? Comment on the controllability of  $(A, Bv)$  pair, where  $v = \begin{bmatrix} 1 \\ 0 \end{bmatrix}$ .

## Solution to Problem 2

### Recall (Lecture Slide 43)

A matrix  $A$  is called *cyclic* whenever the Jordan form of  $A$  has one and only Jordan block associated with each distinct eigenvalue.

Given,

$$A = \begin{bmatrix} -1 & 1 & 0 \\ 3 & 2 & 0 \\ 2 & 0 & 4 \end{bmatrix}$$

The eigenvalues of  $A$  are  $\{4, 2, -1\}$ , forming 3 jordan blocks of size  $(1 \times 1)$ . Therefore,  $A$  is cyclic as the sufficient condition is satisfied.



## Solution to Problem 2

### Recall(Lecture slide 44)

If the  $n$ -dimensional  $p$ -input pair  $(A, B)$  is controllable and if  $A$  is cyclic, then for almost any  $p \times l$  vector  $v$ , the single-input pair  $(A, Bv)$  is controllable.

The controllability matrix of pair  $(A, B)$  is given as

$$\mathfrak{C} = \begin{bmatrix} 2 & 1 & -2 & 1 & 8 & 6 \\ 0 & 2 & 6 & 7 & 6 & 17 \\ 1 & 0 & 8 & 2 & 28 & 10 \end{bmatrix}$$

which has a rank = 3. Since pair  $(A, B)$  is controllable and  $A$  is cyclic, it implies that  $(A, Bv)$  is also controllable.

## Solution to Problem 2

Numerically verifying the claim, we have the controllability matrix of pair  $(A, Bv)$ , where  $v = \begin{bmatrix} 1 \\ 0 \end{bmatrix}$ , equal to

$$\begin{bmatrix} 2 & -2 & 8 \\ 0 & 6 & 6 \\ 1 & 8 & 28 \end{bmatrix}$$

which also has a rank  $= 3$ .

Similarly, this theorem can also be verified for other values of  $v$ .

# State Feedback Design for Multi-Input system

## Problem 3

Let

$$A = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ -3 & 1 & 2 & 3 \\ 2 & 1 & 0 & 0 \end{bmatrix}, \quad B = \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 1 & 2 \\ 0 & 2 \end{bmatrix}$$

Find two different constant matrices  $K$  such that  $(A - BK)$  has eigenvalues  $-4 \pm 3j$  and  $-5 \pm 4j$ .

## Solution to Problem 3

Given,

$$A = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ -3 & 1 & 2 & 3 \\ 2 & 1 & 0 & 0 \end{bmatrix}, \quad B = \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 1 & 2 \\ 0 & 2 \end{bmatrix}$$

Calculating the Controllability matrix of pair  $(A, B)$

$$\mathfrak{C} = \begin{bmatrix} 0 & 0 & 0 & 0 & 1 & 2 & 2 & 10 \\ 0 & 0 & 1 & 2 & 2 & 10 & 5 & 22 \\ 1 & 2 & 2 & 10 & 5 & 22 & 12 & 54 \\ 0 & 2 & 0 & 0 & 1 & 2 & 4 & 14 \end{bmatrix}$$

which has a rank= 4. Therefore the pair  $(A, B)$  is controllable.

## Solution to Problem 3

The Jordan form of  $A$  matrix is then given as

$$\begin{bmatrix} 0.3215 - 1.2581i & 0 & 0 & 0 \\ 0 & 0.3215 + 1.2581i & 0 & 0 \\ 0 & 0 & -1.3262 & 0 \\ 0 & 0 & 0 & 2.6833 \end{bmatrix}$$

Since all the Jordan blocks are of size  $1 \times 1$ ,  $A$  is cyclic. Arbitrarily selecting  $v$  as  $\begin{bmatrix} 1 \\ 2 \end{bmatrix}$ , we can calculate the controllability matrix of pair  $(A, Bv)$  to find that it is also a controllable pair (having rank  $= 4$ ).

We can now proceed forward to design the state feedback controller for the reduced single-input system  $(A, Bv)$ .

## Solution to Problem 3

The state feedback gain vector ( $k$ ) for system defined by  $(A, Bv)$  for eigenvalues  $-4 \pm 3j$  and  $-5 \pm 4j$ , calculated by eigenvalue placement method (equivalent to using `place` command in MATLAB) is

$$[90.7152 \quad 10.5868 \quad 6.0939 \quad -2.6174]$$

## Solution to Problem 3

The state feedback gain vector ( $k$ ) for system defined by  $(A, Bv)$  for eigenvalues  $-4 \pm 3j$  and  $-5 \pm 4j$ , calculated by eigenvalue placement method (equivalent to using `place` command in MATLAB) is

$$[90.7152 \quad 10.5868 \quad 6.0939 \quad -2.6174]$$

Then, the overall state feedback for the multi-input system is  $u(t) = vu'(t)$ , where  $u'(t) = -kx(t)$ .

The gain matrix becomes

$$\begin{aligned} K_1 &= vk = \begin{bmatrix} 1 \\ 2 \end{bmatrix} [90.7152 \quad 10.5868 \quad 6.0939 \quad -2.6174] \\ \implies K_1 &= \begin{bmatrix} 90.7152 & 10.5868 & 6.0939 & -2.6174 \\ 181.4304 & 21.1735 & 12.1878 & -5.2348 \end{bmatrix} \end{aligned}$$

## Solution to Problem 3

Selecting another arbitrary value of  $v$  as  $\begin{bmatrix} 0.8 \\ -1 \end{bmatrix}$ , the state feedback gain vector ( $k$ ) for the same eigenvalues become:

$$[-168.9968 - 62.8041 - 13.2899 - 2.0261]$$

and the multi-input system's gain matrix becomes

$$K_2 = vk = \begin{bmatrix} 0.8 \\ -1 \end{bmatrix} [-168.9968 - 62.8041 - 13.2899 - 2.0261]$$
$$\Rightarrow K_2 = \begin{bmatrix} -135.1975 & -50.2433 & -10.6319 & -1.6209 \\ 168.9968 & 62.8041 & 13.2899 & 2.0261 \end{bmatrix}$$



## Problem 4

Consider the system

$$A = \begin{bmatrix} -10 & 1 \\ -0.02 & -2 \end{bmatrix}, \quad b = \begin{bmatrix} 0 \\ 2 \end{bmatrix}, \quad c = [1 \quad 0]$$

- a) Design a controller such that the desired eigenvalues are located at  $s = -5 \pm j$  and the output tracks a unit step input, i.e.  $r = 1$ .
- b) Plot the step response of the system under the effect of an external step disturbance ( $w$ ) as in

$$\dot{x} = Ax + bu + bw$$

# Solution to Problem 4

- a) For the system

$$A = \begin{bmatrix} -10 & 1 \\ -0.02 & -2 \end{bmatrix}, \quad b = \begin{bmatrix} 0 \\ 2 \end{bmatrix}, \quad c = [1 \quad 0]$$

Firstly, we calculate the desired feedback gains for eigenvalues at  $s = -5 \pm j$ .

## Solution to Problem 4

- (a) For the system

$$A = \begin{bmatrix} -10 & 1 \\ -0.02 & -2 \end{bmatrix}, \quad b = \begin{bmatrix} 0 \\ 2 \end{bmatrix}, \quad c = [1 \quad 0]$$

Firstly, we calculate the desired feedback gains for eigenvalues at  $s = -5 \pm j$ .

The characteristic equation of the closed-loop state feedback system becomes

$$\begin{aligned} |sI - A + bk| &= 0 \\ \left| \begin{bmatrix} s+10 & -1 \\ 0.02 & s+2 \end{bmatrix} + \begin{bmatrix} 0 \\ 2 \end{bmatrix} \begin{bmatrix} k_1 & k_2 \end{bmatrix} \right| &= 0 \\ \begin{vmatrix} s+10 & -1 \\ 0.02+2k_1 & s+2+2k_2 \end{vmatrix} &= 0 \\ \Rightarrow s^2 + s(12+2k_2) + 19.98 + 20k_2 - 2k_1 &= 0 \end{aligned}$$

# Solution to Problem 4

- a) For the system

$$A = \begin{bmatrix} -10 & 1 \\ -0.02 & -2 \end{bmatrix}, \quad b = \begin{bmatrix} 0 \\ 2 \end{bmatrix}, \quad c = \begin{bmatrix} 1 & 0 \end{bmatrix}$$

Firstly, we calculate the desired feedback gains for eigenvalues at  $s = -5 \pm j$ .

The characteristic equation of the closed-loop state feedback system becomes

$$\begin{aligned} |sI - A + bk| &= 0 \\ \left| \begin{bmatrix} s + 10 & -1 \\ 0.02 & s + 2 \end{bmatrix} + \begin{bmatrix} 0 \\ 2 \end{bmatrix} \begin{bmatrix} k_1 & k_2 \end{bmatrix} \right| &= 0 \\ \left| \begin{array}{cc} s + 10 & -1 \\ 0.02 + 2k_1 & s + 2 + 2k_2 \end{array} \right| &= 0 \\ \implies s^2 + s(12 + 2k_2) + 19.98 + 20k_2 - 2k_1 &= 0 \end{aligned}$$

The desired characteristic equation is

$$\begin{aligned} (s + 5 + j)(s + 5 - j) &= 0 \\ s^2 + 10s + 26 &= 0 \end{aligned}$$

## Solution to Problem 4

- ④ Comparing the desired and actual characteristic equations, we get

$$k = [12.99 \quad -1]$$

# Solution to Problem 4

- Ⓢ Comparing the desired and actual characteristic equations, we get

$$k = [12.99 \quad -1]$$

Now, the transfer function of the closed-loop system becomes

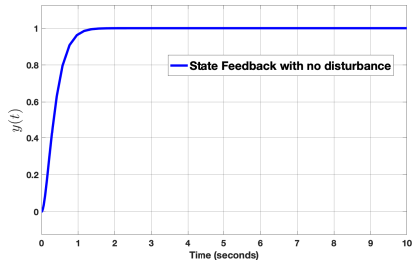
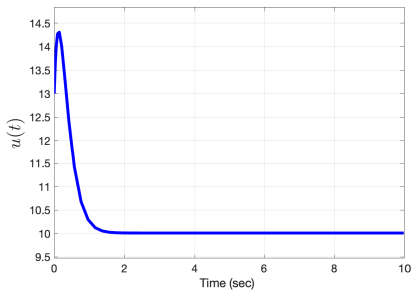
$$\begin{aligned}\hat{g}(s) &= c(sI - A - bk)^{-1}b \\ &= [1 \quad 0] \left( \begin{bmatrix} s+10 & -1 \\ 0.02 & s+2 \end{bmatrix} - \begin{bmatrix} 0 \\ 2 \end{bmatrix} [12.99 \quad -1] \right)^{-1} \begin{bmatrix} 0 \\ 2 \end{bmatrix} \\ \Rightarrow \hat{g}(s) &= \frac{2}{s(s-10) + 26}\end{aligned}$$

The steady state value of output  $y(t) = \hat{g}(0) = \frac{1}{13}$ .

Since  $\hat{g}(0) \neq (r(t) = 1)$ , we will use  $p$  (feedforward gain) equal to  $\frac{1}{\hat{g}(0)} = 13$  to make the output track the unit step input ( $r = 1$ ).

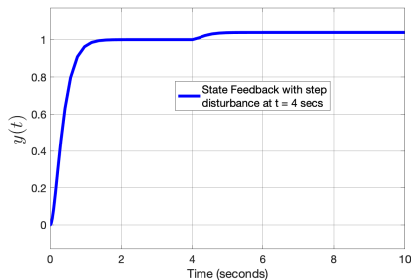
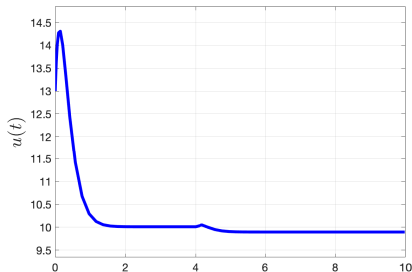
# Solution to Problem 4

Consequently, the unit step response becomes:



## Solution to Problem 4

- b)** Adding a step disturbance at  $t = 4$  secs to the system, the step response becomes



It is evident from the step response that the state feedback control is unable to track the reference( $r = 1$ ) due to the effect of this external disturbance.

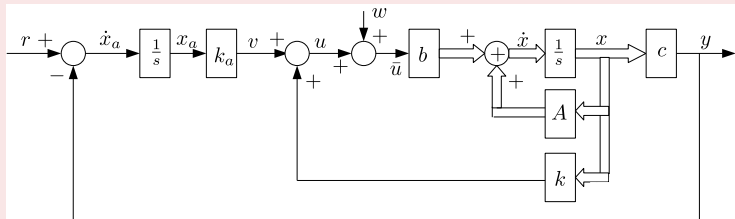


## Problem 4, continued

- ④ Design a robust controller to reject the effect of disturbances.

# Solution to Problem 4

## Recall (Lecture Slides 29)



In the above closed-loop control design, the output  $y$  will track asymptotically and robustly any step reference input  $r(t) = a$  and reject any step disturbance with unknown magnitude.

## Solution to Problem 4

- Adding an integral control action to the closed-loop system, the  $(A, b)$  pair becomes,

$$\bar{A} = \begin{bmatrix} A & 0 \\ -c & 0 \end{bmatrix}, \quad \bar{B} = \begin{bmatrix} b \\ 0 \end{bmatrix}$$

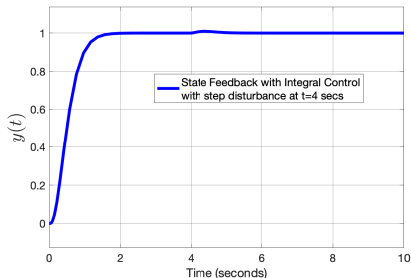
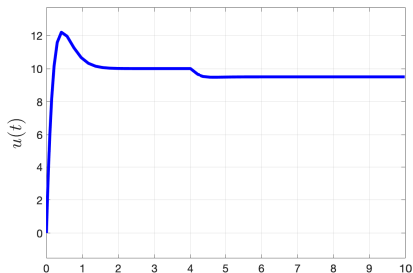
Again, calculating the feedback gain vector  $k_{fb} = [k \quad k_a]$ , we get

$$[12.99 \quad 2 \quad -78]$$

where the last element represents the integral gain  $k_a$  of the controller.

# Solution to Problem 4

Ⓢ The step response now becomes,



which clearly shows that the state feedback control with integral action is capable of disturbance rejection.

## Problem 5

Consider the non-linear system

$$\dot{x} = f(x, u), \quad x \in \mathbb{R}^n, u \in \mathbb{R}^k \quad (\text{NLS})$$

and continuously differentiable function  $V : \mathbb{R}^n \rightarrow \mathbb{R}$  with  $V(0) = 0$ . Verify that the functional

$$H(x(\cdot), u(\cdot)) \triangleq - \int_0^\infty \frac{\partial V(x(t))}{\partial x} f(x(t), u(t)) dt$$

is a feedback invariant as long as  $\lim_{t \rightarrow \infty} x(t) = 0$ .

---

<sup>1</sup>Hespanha Exercise 20.1

# Solution to Problem 5

## Recall!

Recall from the lecture slide 56 , that a functional  $H(x(\cdot), u(\cdot))$  that involves system's input and state is a feedback invariant for a given dynamical system if when computed along a solution to the system, its value depends only on the initial condition and not on the specific input signal.

# Solution to Problem 5

## Recall!

Recall from the lecture slide 56 , that a functional  $H(x(\cdot), u(\cdot))$  that involves system's input and state is a feedback invariant for a given dynamical system if when computed along a solution to the system, its value depends only on the initial condition and not on the specific input signal.

Given

$$H(x(\cdot), u(\cdot)) = - \int_0^{\infty} \frac{\partial V(x(t))}{\partial x} f(x(t), u(t)) dt$$

# Solution to Problem 5

## Recall!

Recall from the lecture slide 56 , that a functional  $H(x(\cdot), u(\cdot))$  that involves system's input and state is a feedback invariant for a given dynamical system if when computed along a solution to the system, its value depends only on the initial condition and not on the specific input signal.

Given

$$H(x(\cdot), u(\cdot)) = - \int_0^\infty \frac{\partial V(x(t))}{\partial x} f(x(t), u(t)) dt$$

The derivative of  $V$  along the trajectories of (NLS) denoted by  $\dot{V}(x)$ , is given by

$$\begin{aligned} \dot{V}(x) &= \sum_{i=1}^n \frac{\partial V}{\partial x_i} \dot{x}_i = \sum_{i=1}^n \frac{\partial V}{\partial x_i} f_i(x) \\ &= \begin{bmatrix} \frac{\partial V}{\partial x_1} & \frac{\partial V}{\partial x_2} & \cdots & \frac{\partial V}{\partial x_n} \end{bmatrix} \begin{bmatrix} f_1(x) \\ f_2(x) \\ \vdots \\ f_n(x) \end{bmatrix} = \frac{\partial V}{\partial x} f(x) \end{aligned}$$

The derivative of  $V$  along the trajectories of a system is dependent on the system's equation.



## Solution to Problem 5

Consider

$$H(x(\cdot), u(\cdot)) = - \int_0^\infty \frac{\partial V(x(t))}{\partial x} f(x(t), u(t)) dt$$

## Solution to Problem 5

Consider

$$\begin{aligned} H(x(\cdot), u(\cdot)) &= - \int_0^\infty \frac{\partial V(x(t))}{\partial x} f(x(t), u(t)) dt \\ &= - \int_0^\infty \dot{V}(x(t)) dt \\ &= -(V(x(\infty)) - V(x(0))) \\ &= V(x(0)) \quad \text{as long as } \lim_{t \rightarrow \infty} x(t) = 0 \end{aligned}$$

Since  $H(x(\cdot), u(\cdot))$  depends only on the initial state of the system, it is a feedback invariant.

# Linear Quadratic Regulator(LQR)

## Problem 6

Given a system

$$\begin{aligned}\dot{x}_1(t) &= x_2(t), & x_1(0) &= 5 \\ \dot{x}_2(t) &= -2x_1(t) + 5x_2(t) + u(t), & x_2(0) &= 10\end{aligned}$$

and the performance index (PI)

$$J = \frac{1}{2} \int_0^{\infty} [2x_1^2(t) + 6x_1(t)x_2(t) + 5x_2^2(t) + 0.25u^2(t)] dt$$

obtain the feedback control law. Compare the performance for different input and state weighting matrices.

---

<sup>1</sup>Naidu, Optimal Control Systems, Example 3.1

## Solution to Problem 6

From the given system and performance index, the various quantities are

$$A = \begin{bmatrix} 0 & 1 \\ -2 & 5 \end{bmatrix}; \quad b = \begin{bmatrix} 0 \\ 1 \end{bmatrix}; \quad Q = \begin{bmatrix} 2 & 3 \\ 3 & 5 \end{bmatrix}; \quad r = \frac{1}{4}; \quad t_0 = 0.$$

It is easy to check that the system is unstable.

## Solution to Problem 6

From the given system and performance index, the various quantities are

$$A = \begin{bmatrix} 0 & 1 \\ -2 & 5 \end{bmatrix}; \quad b = \begin{bmatrix} 0 \\ 1 \end{bmatrix}; \quad Q = \begin{bmatrix} 2 & 3 \\ 3 & 5 \end{bmatrix}; \quad r = \frac{1}{4}; \quad t_0 = 0.$$

It is easy to check that the system is unstable. Let  $P$  be the  $2 \times 2$  symmetric matrix

$$P = \begin{bmatrix} p_{11} & p_{12} \\ p_{21} & p_{22} \end{bmatrix}$$

Then, the optimal control is given by

$$u^* = -r^{-1}b'Px^*$$

where  $P$  is the solution of the algebraic Riccati equation

$$A'P + PA + Q - Pbr^{-1}b'P = 0$$

# Solution to Problem 6

From the given system and performance index, the various quantities are

$$A = \begin{bmatrix} 0 & 1 \\ -2 & 5 \end{bmatrix}; \quad b = \begin{bmatrix} 0 \\ 1 \end{bmatrix}; \quad Q = \begin{bmatrix} 2 & 3 \\ 3 & 5 \end{bmatrix}; \quad r = \frac{1}{4}; \quad t_0 = 0.$$

It is easy to check that the system is unstable. Let  $P$  be the  $2 \times 2$  symmetric matrix

$$P = \begin{bmatrix} p_{11} & p_{12} \\ p_{21} & p_{22} \end{bmatrix}$$

Then, the optimal control is given by

$$u^* = -r^{-1}b'Px^*$$

where  $P$  is the solution of the algebraic Riccati equation

$$A'P + PA + Q - Pbr^{-1}b'P = 0$$

This equation can be solved using the `care` command in MATLAB.

# Solution to Problem 6

The next simulation results show the variation of the trajectories for different weighting matrices

①  $Q_1 = Q = \begin{bmatrix} 2 & 3 \\ 3 & 5 \end{bmatrix}, r_1 = r = 0.25$  and

②  $Q_2 = 4Q$  and  $r_2 = r = 0.25$ .

# Solution to Problem 6

## Simulations

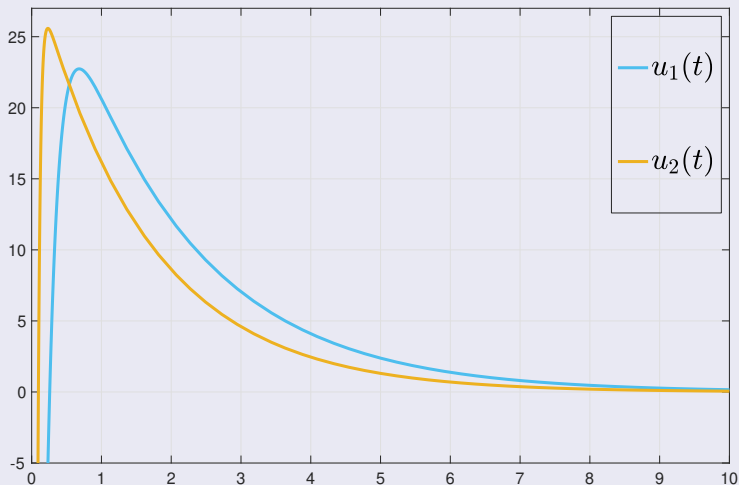


Figure: Input trajectories ( $u_1$  for  $Q_1$  and  $r_1$  and  $u_2$  for  $Q_2$  and  $r_2$ )



# Solution to Problem 6

## Simulations

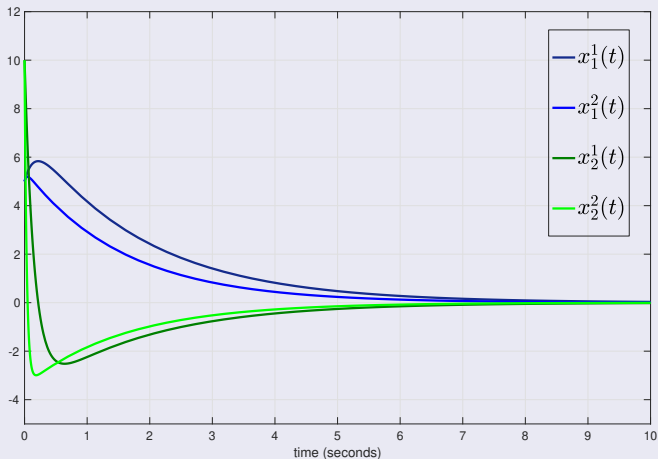


Figure: State and trajectories ( $x_1^1$  and  $x_2^1$  for  $Q_1$ ,  $r_1$  and  $x_1^2$  and  $x_2^2$  for  $Q_2$  and  $r_2$ )