

Linear Dynamical Systems

Week 5 - State Feedback Controller Design

- ① Open-loop control
- ② State-feedback controller design
- ③ Regulation and Tracking
- ④ Extension to Multivariable case
- ⑤ Preview of Optimal control

Control Problem

Open-loop minimum-energy control

Suppose that a particular state x_1 belongs to the reachable subspace $\mathcal{R}[t_0, t_1]$ of the system (AB-CLTV).

Theorem (Reachable subspace)

Given two times $t_1 > t_0 \geq 0$,

$$\mathcal{R}[t_0, t_1] = \text{Im}W_R(t_0, t_1).$$

Moreover, if $x_1 = W_R(t_0, t_1)\eta_1 \in \text{Im}W_R(t_0, t_1)$, the control

$$u(t) = B(t)^T \phi(t_1, t)^T \eta_1, \quad t \in [t_0, t_1] \quad (\text{Min-energy control})$$

can be used to transfer the state from $x(t_0) = 0$ to $x(t_1) = x_1$.

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can be used to transfer the state from $x(t_0) = 0$ to $x(t_1) = x_1$.

In general, there may be other control that achieve the first goal, but controls of the form (Min-energy control) are desirable because they *minimize control energy*.

Open-loop minimum-energy control

Suppose that $\bar{u}(\cdot)$ is another control that transfers the state to x_1 and therefore

$$x_1 = \int_{t_0}^{t_1} \phi(t_1, \tau) B(\tau) u(\tau) d\tau = \int_{t_0}^{t_1} \phi(t_1, \tau) B(\tau) \bar{u}(\tau) d\tau.$$

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For this to hold, we must have

$$\int_{t_0}^{t_1} \phi(t_1, \tau) B(\tau) v(\tau) d\tau = 0$$

where $v = \bar{u} - u$.

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where $v = \bar{u} - u$. The “energy” of $\bar{u}(\cdot)$ can be related to the energy of $u(\cdot)$ as follows

$$\int_{t_0}^{t_1} \|\bar{u}(\tau)\|^2 d\tau = \int_{t_0}^{t_1} \|\overbrace{B'(t)\phi'(t_1, \tau)\eta_1}^{u(\tau)} + v(\tau)\|^2 d\tau$$

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$$\begin{aligned} \int_{t_0}^{t_1} \|\bar{u}(\tau)\|^2 d\tau &= \int_{t_0}^{t_1} \|\overbrace{B'(t)\phi'(t_1, \tau)\eta_1}^{u(\tau)} + v(\tau)\|^2 d\tau \\ &= \eta_1' W_R(t_0, t_1) \eta_1 + \int_{t_0}^{t_1} \|v(\tau)\|^2 d\tau + 2\eta_1' \int_{t_0}^{t_1} \phi(t_1, \tau) B(\tau) v(\tau) d\tau \end{aligned}$$

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Note the last term is equal to zero, and we conclude that the energy of \bar{u} is minimized for $v(\cdot) = 0$, i.e., for $\bar{u} = u$.

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For this to hold, we must have

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where $v = \bar{u} - u$. The “energy” of $\bar{u}(\cdot)$ can be related to the energy of $u(\cdot)$ as follows

$$\begin{aligned} \int_{t_0}^{t_1} \|\bar{u}(\tau)\|^2 d\tau &= \int_{t_0}^{t_1} \|\overbrace{B'(t)\phi'(t_1, \tau)\eta_1}^{u(\tau)} + v(\tau)\|^2 d\tau \\ &= \eta_1' W_R(t_0, t_1) \eta_1 + \int_{t_0}^{t_1} \|v(\tau)\|^2 d\tau + 2\eta_1' \int_{t_0}^{t_1} \phi(t_1, \tau) B(\tau) v(\tau) d\tau \end{aligned}$$

Note the last term is equal to zero, and we conclude that the energy of \bar{u} is minimized for $v(\cdot) = 0$, i.e., for $\bar{u} = u$. Moreover, for $v(\cdot) = 0$, we conclude that the energy required for the optimal control $u(\cdot)$ in (Min-energy control) is given by

$$\int_{t_0}^{t_1} \|u(\tau)\|^2 d\tau = \eta_1' W_R(t_0, t_1) \eta_1.$$

Open-loop minimum-energy control

Theorem (Reachable and Controllable subspaces)

- 1 if $x_1 = W_R(t_0, t_1)\eta_1 \in \text{Im}W_R(t_0, t_1)$, the control

$$u(t) = B(t)^T \phi(t_1, t)^T \eta_1, \quad t \in [t_0, t_1] \quad (1)$$

can be used to transfer the state from $x(t_0) = 0$ to $x(t_1) = x_1$.

- 2 if $x_0 = W_C(t_0, t_1)\eta_0 \in \text{Im}W_C(t_0, t_1)$, the control

$$u(t) = -B(t)^T \phi(t_0, t)^T \eta_0, \quad t \in [t_0, t_1] \quad (2)$$

can be used to transfer the state $x(t_0) = x_0$ to $x(t_1) = 0$.

Open-loop minimum-energy control

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$$u(t) = -B(t)^T \phi(t_0, t)^T \eta_0, \quad t \in [t_0, t_1] \quad (2)$$

can be used to transfer the state $x(t_0) = x_0$ to $x(t_1) = 0$.

Theorem (Minimum-energy control)

Given two times $t_1 > t_0 \geq 0$,

- 1 when $x_1 \in \mathcal{R}[t_0, t_1]$, the control (1) transfers the state from $x(t_0) = 0$ to $x(t_1) = x_1$ with the smallest amount of control energy, which is given by

$$\int_{t_0}^{t_1} \|u(\tau)\|^2 d\tau = \eta_1' W_R(t_0, t_1) \eta_1,$$

- 2 when $x_1 \in \mathcal{C}[t_0, t_1]$, the control (2) transfers the state from $x(t_0) = x_0$ to $x(t_1) = 0$ with the smallest amount of control energy, which is given by

$$\int_{t_0}^{t_1} \|u(\tau)\|^2 d\tau = \eta_0' W_C(t_0, t_1) \eta_0.$$

Solution by Inversion

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State Feedback

Consider the n -dimensional *single-variable* state equation

$$\begin{aligned}\dot{x} &= Ax + bu \\ y &= Cx\end{aligned}\tag{LTI}$$

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$$u = r - Kx = r - [k_1 \quad k_2 \quad \dots \quad k_n] x = r - \sum_{i=1}^n k_i x_i.\tag{3}$$

Each feedback gain k_i is a real constant. This is called the *constant gain negative state feedback* or, simply, *state feedback*.

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$$\begin{aligned}\dot{x} &= (A - bk)x + br \\ y &= cx\end{aligned}\tag{CL-LTI}$$

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$$\begin{aligned}\dot{x} &= (A - bk)x + br \\ y &= cx\end{aligned}\tag{CL-LTI}$$

Theorem

The pair $(A - bk, b)$, for any $1 \times n$ real constant vector k , is controllable if and only if (A, b) is controllable.

State Feedback

Proof.

We show the theorem for $n = 4$. Define

$$\mathfrak{C} = [b \quad Ab \quad A^2b \quad A^3b]$$

and

$$\mathfrak{C}_f = [b \quad (A - bk)b \quad (A - bk)^2b \quad (A - bk)^3b]$$

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$$\mathfrak{C}_f = [b \quad (A - bk)b \quad (A - bk)^2b \quad (A - bk)^3b]$$

It is straightforward to verify

$$\mathfrak{C}_f = \mathfrak{C} \begin{bmatrix} 1 & -kb & -k(A - bk)b & -k(A - bk)^2b \\ 0 & 1 & -kb & -k(A - bk)b \\ 0 & 0 & 1 & -kb \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

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Note that k is $1 \times n$ and b is $n \times 1$. Thus kb is scalar; so is every entry in the rightmost matrix. Because the right most matrix is nonsingular for any k , the rank of \mathfrak{C}_f equals the rank of \mathfrak{C} . Thus (CL-LTI) is controllable if and only if (LTI) is controllable. \square

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Note

The input r does not control the state x directly; it generates u to control x . Therefore, if u cannot control x , neither can r .

State Feedback

Theorem

Consider the (LTI) system with $n = 4$ and the characteristic polynomial

$$\Delta(s) = \det(sI - A) = s^4 + \alpha_1 s^3 + \alpha_2 s^2 + \alpha_3 s + \alpha_4$$

If the system is controllable, then it can be transformed by the transformation $\bar{x} = Px$ with

$$Q := P^{-1} = [b \quad Ab \quad A^2b \quad A^3b] \begin{bmatrix} 1 & \alpha_1 & \alpha_2 & \alpha_3 \\ 0 & 1 & \alpha_1 & \alpha_2 \\ 0 & 0 & 1 & \alpha_1 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

into the controllable canonical form

$$\dot{\bar{x}} = \bar{A}\bar{x} + \bar{b}u = \begin{bmatrix} -\alpha_1 & -\alpha_2 & -\alpha_3 & -\alpha_4 \\ 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \bar{x} + \begin{bmatrix} 1 \\ 0 \\ 0 \\ 0 \end{bmatrix} u$$

$$y = \bar{c}\bar{x} = [\beta_1 \quad \beta_2 \quad \beta_3 \quad \beta_4] \bar{x}$$

Furthermore, the transfer function of the system with $n = 4$ equals

$$\hat{g}(s) = \frac{\beta_1 s^3 + \beta_2 s^2 + \beta_3 s + \beta_4}{s^4 + \alpha_1 s^3 + \alpha_2 s^2 + \alpha_3 s + \alpha_4}$$

State Feedback

Theorem (Eigenvalue Assignment)

*If the n -dimensional (LTI) system is controllable, then by state feedback $u = r - kx$, where k is a $1 \times n$ real constant vector, the eigenvalues of $A - bk$ can **arbitrarily** be assigned provided that complex conjugate eigenvalues are assigned in pairs.*

Proof

Let $n = 4$, if (LTI) is controllable then it can be transformed into the CCF $\dot{\bar{x}} = \bar{A}\bar{x} + \bar{b}\bar{u}$ where $\bar{A} = PAP^{-1}$, $\bar{b} = Pb$, and $\bar{c} = P\mathcal{C}$.

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Substituting $\bar{x} = Px$ in u yields

$$u = r - kx = r - kP^{-1}\bar{x} = r - \bar{k}\bar{x}.$$

Since $\bar{A} - \bar{b}\bar{k} = P(A - bk)P^{-1}$, it implies $\lambda[A - bk] = \lambda[\bar{A} - \bar{b}\bar{k}]$.

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From any set of desired eigenvalues, we can form

$$\Delta_f(s) = s^4 + \bar{\alpha}_1 s^3 + \bar{\alpha}_2 s^2 + \bar{\alpha}_3 s + \bar{\alpha}_4.$$

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$$\bar{k} = [\bar{\alpha}_1 - \alpha_1 \quad \bar{\alpha}_2 - \alpha_2 \quad \bar{\alpha}_3 - \alpha_3 \quad \bar{\alpha}_4 - \alpha_4]$$

the state feedback equation becomes

$$\dot{\bar{x}} = (\bar{A} - \bar{b}\bar{k})\bar{x} + \bar{b}r = \begin{bmatrix} -\bar{\alpha}_1 & -\bar{\alpha}_2 & -\bar{\alpha}_3 & -\bar{\alpha}_4 \\ 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \bar{x} + \begin{bmatrix} 1 \\ 0 \\ 0 \\ 0 \end{bmatrix} r$$

$$y = [\beta_1 \quad \beta_2 \quad \beta_3 \quad \beta_4] \bar{x}$$

Proof

Let $n = 4$, if (LTI) is controllable then it can be transformed into the CCF $\dot{\bar{x}} = \bar{A}\bar{x} + \bar{b}\bar{u}$ where $\bar{A} = PAP^{-1}$, $\bar{b} = Pb$, and $\bar{c} = P\mathcal{C}$.

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Because of the companion form, the characteristic polynomial of $(\bar{A} - \bar{b}\bar{k})$ and of $(A - bk)$ equals $\Delta_f(s)$. Thus the state feedback equation has the set of desired eigenvalues. The feedback gain k can be computed from

$$k = \bar{k}P = \bar{k}\bar{c}\mathcal{C}^{-1}.$$

Feedback transfer function

$$(A, b, c) \implies \hat{g}(s) = c(sI - A)^{-1}b = \frac{\beta_1 s^3 + \beta_2 s^2 + \beta_3 s + \beta_4}{s^4 + \alpha_1 s^3 + \alpha_2 s^2 + \alpha_3 s + \alpha_4}$$

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After state feedback

$$(A - bk, b, c) \implies \hat{g}_f(s) = c(sI - A + bk)^{-1}b = \frac{\beta_1 s^3 + \beta_2 s^2 + \beta_3 s + \beta_4}{s^4 + \bar{\alpha}_1 s^3 + \bar{\alpha}_2 s^2 + \bar{\alpha}_3 s + \bar{\alpha}_4}$$

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Note

- the numerators are the same, state feedback can shift the poles of a plant but has *no effect on the zeros*,
- state feedback *may alter the observability* property because one or more poles are shifted to coincide with the zeros of $\hat{g}(s)$.

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Attention

The command `K=place(A,B,v)` computes a matrix K such that the eigenvalues of $A - BK$ are those specified in the vector v . The pair (A, B) should be controllable, and the vector v should have no repeated eigenvalues. This command should be used with great caution (and generally avoided), because it is numerically badly conditioned.

Selection of desired eigenvalues¹

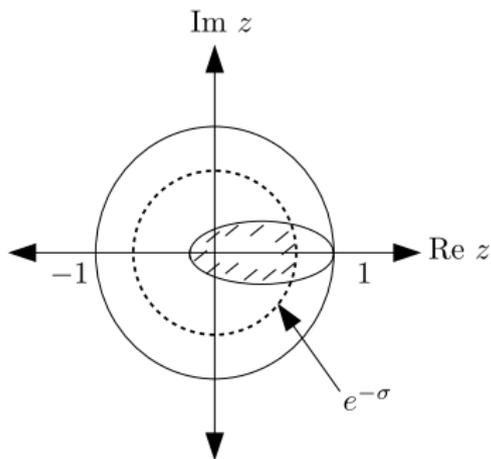
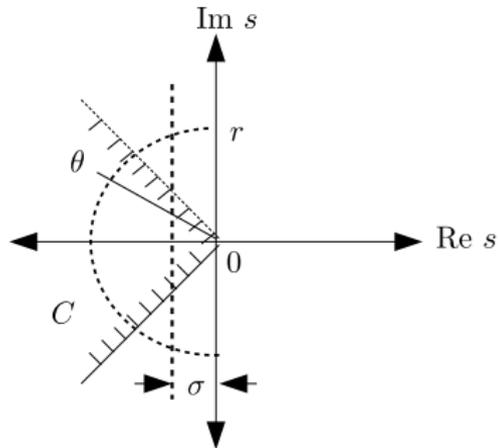
- ① depends on the performance criteria
 - rise time
 - overshoot
 - settling time
- ② response depends upon the poles and zeros both
- ③ factors affecting the selection of poles
 - zeros of the plant
 - magnitude of u : saturation or burn out
 - rise time, settling time, overshoot
 - bandwidth of the closed-loop
- ④ involve compromises among many conflicting objectives

¹Boyd *et al.* Linear controller design: limits of performance. Englewood Cliffs, NJ: Prentice Hall, 1991.

Some guidelines

As a guide, place all the poles inside the region denoted by C

- larger the σ , faster the response
- large the θ , larger the overshoot
- larger the r , faster the response, u will also be larger, BW will also be larger and the resulting system will be more susceptible to noise



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^aOnce F and \bar{k} are selected, we may use the MATLAB function `lyap` to solve the Lyapunov equation

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What remains to be proved is the nonsingularity of T !

Nonsingularity of T

Theorem

If A and F have no eigenvalues in common, then the unique solution T of $AT - TF = b\bar{k}$ is nonsingular if and only if (A, b) and (F', \bar{k}') are controllable pairs.

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If A and F have no eigenvalues in common, then the unique solution T of $AT - TF = \bar{b}\bar{k}$ is nonsingular if and only if (A, b) and (F', \bar{k}') are controllable pairs.

We shall prove the theorem for $n = 4$.

Recall, the characteristic polynomial of A is given by

$$\Delta(s) = s^4 + \alpha_1 s^3 + \alpha_2 s^2 + \alpha_3 s + \alpha_4$$

then from Cayley-Hamilton theorem we have

$$\Delta(A) = A^4 + \alpha_1 A^3 + \alpha_2 A^2 + \alpha_3 A + \alpha_4 I = 0$$

Proof

Let us consider

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$$\det \Delta(F) = \prod_i \Delta(\bar{\lambda}_i) \neq 0$$

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$$\begin{aligned} A^2T - TF^2 &= A(TF + b\bar{k}) - TF^2 = Ab\bar{k} + (AT - TF)F \\ &= Ab\bar{k} + b\bar{k}F \end{aligned}$$

Proof (Cont...)

Proceeding forward, we can obtain the following set of equations:

$$IT - TI = 0$$

$$AT - TF = b\bar{k}$$

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If (A, b) and (F', \bar{k}') are controllable, then all three matrices are nonsingular, which implies that T is nonsingular.

If (A, b) and/or (F, \bar{k}) are uncontrollable, then the product of the three matrices is singular. Therefore T is singular. This establishes the theorem.

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- If the desired eigenvalues are all distinct, we can also use the modal form. For example, if $n = 5$, and if the five distinct desired eigenvalues are selected as $\lambda_1, \alpha_1 \pm j\beta_1$ and $\alpha_2 \pm j\beta_2$, then we can select F as

$$F = \begin{bmatrix} \lambda_1 & 0 & 0 & 0 & 0 \\ 0 & \alpha_1 & \beta_1 & 0 & 0 \\ 0 & -\beta_1 & \alpha_1 & 0 & 0 \\ 0 & 0 & 0 & \alpha_2 & \beta_2 \\ 0 & 0 & 0 & -\beta_2 & \alpha_2 \end{bmatrix}$$

It is a block-diagonal matrix.

Regulation

Regulation problem

Suppose the reference signal r is zero, and the response of the system is caused by some nonzero initial conditions. The problem is to find a state feedback gain so that the response will die out at a desired rate.

Examples:

- Aircraft cruise control
- Liquid level control in tanks

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Consider a plant described by (A, b, c) . If A is unstable, then the response excited by any nonzero initial conditions will grow unbounded.

Let $u = r - kx$. Then the state feedback equation becomes $(A - bk, b, c)$ and the response caused by $x(0)$ is

$$y(t) = ce^{(A-bk)t}x(0)$$

Tracking

Tracking problem

Suppose the reference signal r is a constant or $r(t) = a$, for $t \geq 0$. The problem is to design an overall system so that $y(t)$ approaches $r(t) = a$ as t approaches infinity. This is called *asymptotic tracking* of a step reference input.

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Why do we then study these two problems separately?

A linear state equation is often obtained by shifting an operating point and linearization, and the equation is valid only for r very small or zero.

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Tracking a non-constant reference signal is called a *servomechanism* problem and is a much more difficult problem.

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$$(A, b, c) \implies \hat{g}(s) = c(sI - A)^{-1}b = \frac{\beta_1 s^3 + \beta_2 s^2 + \beta_3 s + \beta_4}{s^4 + \alpha_1 s^3 + \alpha_2 s^2 + \alpha_3 s + \alpha_4}$$

After the state feedback and feedforward, it will now become

$$\hat{g}_f(s) = \frac{\hat{y}(s)}{\hat{r}(s)} = p \frac{\beta_1 s^3 + \beta_2 s^2 + \beta_3 s + \beta_4}{s^4 + \bar{\alpha}_1 s^3 + \bar{\alpha}_2 s^2 + \bar{\alpha}_3 s + \bar{\alpha}_4}$$

If (A, b) is controllable, all eigenvalues of $(A - bk)$ or, equivalently, all poles of $\hat{g}_f(s)$ can be assigned arbitrarily. Under this assumption, if the reference input is a step function with magnitude a , then the output $y(t)$ will approach the constant $\hat{g}_f(0).a$ as $t \rightarrow \infty$. Thus in order for $y(t)$ to track asymptotically any step reference input, we need

$$1 = \hat{g}_f(0) = p \frac{\beta_4}{\bar{\alpha}_4} \quad \text{or} \quad p = \frac{\bar{\alpha}_4}{\beta_4}$$

which requires $\beta_4 \neq 0$, which is possible **if and only if the plant transfer function $\hat{g}(s)$ has no zero at $s = 0$.**

Robust Tracking and Disturbance Rejection

- 1 The state equation and transfer function developed to describe a plant may change due to change of load , environment or aging. Thus plant parameter variations often occur in practice.

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a different controller design that can achieve robust tracking and disturbance rejection.

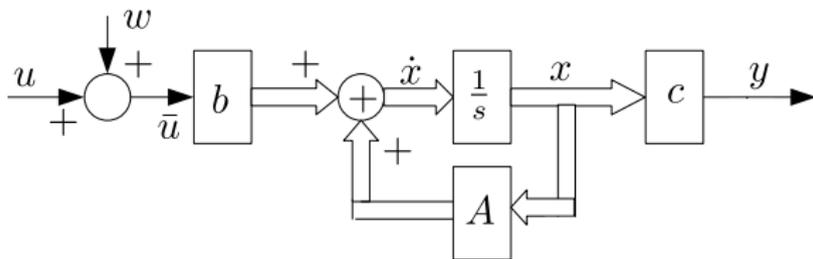
Robust Tracking and Disturbance Rejection

Control problem

Consider a plant described by (LTI) affected by a constant disturbance w with “unknown magnitude” enters at the plant input. Then the state equation is given as

$$\dot{x} = Ax + bu + bw, \quad y = cx \quad (\text{Disturbed LTI})$$

The problem is to design an overall system so that the output $y(t)$ will track asymptotically any step reference input even with the presence of disturbance $w(t)$ and with “plant parameter variations”.

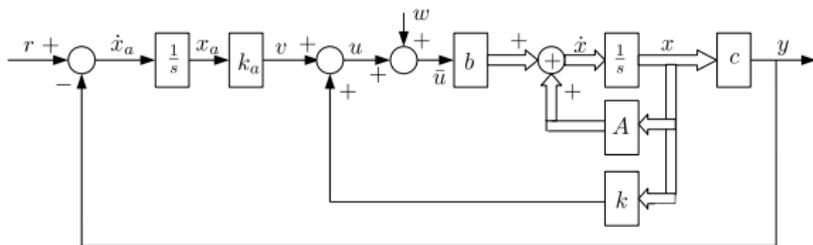


Robust Tracking and Disturbance Rejection

In order to solve this problem, in addition to introducing state feedback, we will introduce an integrator and a unity feedback from the output.

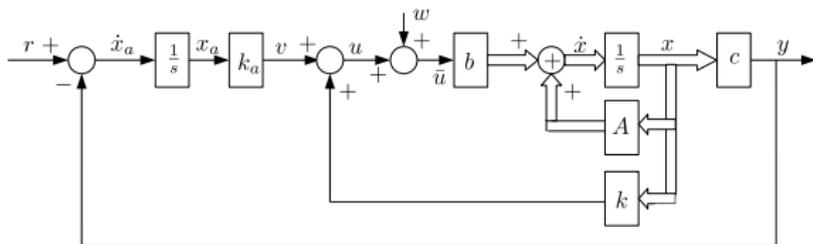
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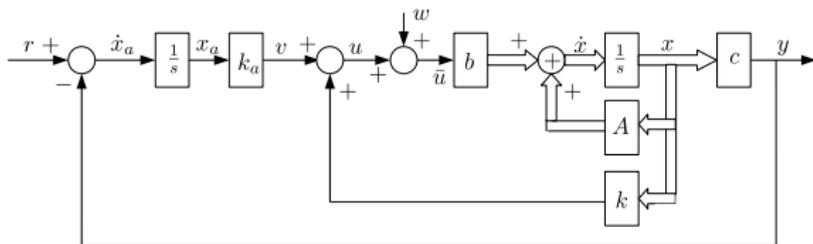
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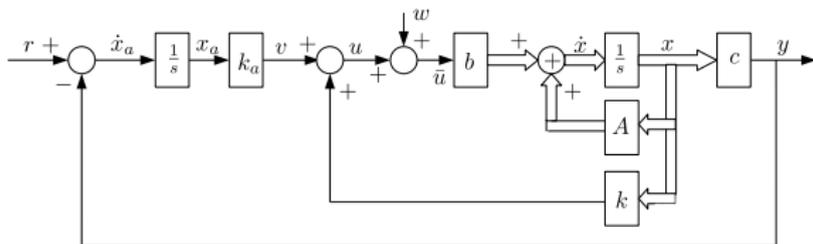


Let the output of the integrator be denoted by $x_a(t)$, an augmented state variable. Then the system has the augmented state vector $\text{col}(x, x_a)$. We now have

$$\dot{x}_a = r - y = r - cx$$

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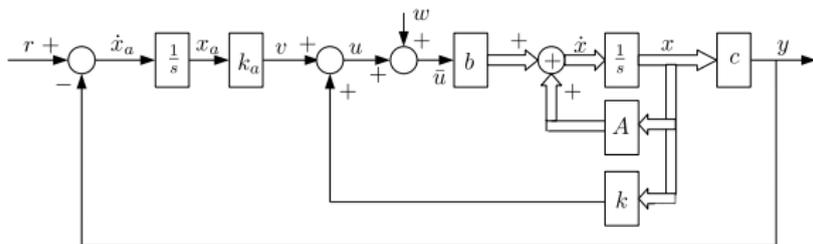
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Substituting these into (Disturbed LTI) yields

$$\begin{bmatrix} \dot{x} \\ \dot{x}_a \end{bmatrix} = \begin{bmatrix} A + bk & bk_a \\ -c & 0 \end{bmatrix} \begin{bmatrix} x \\ x_a \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} r + \begin{bmatrix} b \\ 0 \end{bmatrix} w$$

$$y = \begin{bmatrix} c & 0 \end{bmatrix} \begin{bmatrix} x \\ x_a \end{bmatrix}$$

(Aug. Dist. CL-LTI)

Robust Tracking and Disturbance Rejection

Theorem (Lemma)

If (A, b) is controllable and if $\hat{g}(s) = c(sI - A)^{-1}b$ has no zero at $s = 0$, then all eigenvalues of the “new” A -matrix can be assigned arbitrarily by selecting a feedback gain $[k \quad k_a]$.

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We show the theorem for $n = 4$.

Assumption

- ① A, b and c have been transformed into controllable canonical form.
- ② The transfer function has no zeros at $s = 0$ if and only if $\beta_4 \neq 0$.

$$(A, b, c) \implies \hat{g}(s) = c(sI - A)^{-1}b = \frac{\beta_1 s^3 + \beta_2 s^2 + \beta_3 s + \beta_4}{s^4 + \alpha_1 s^3 + \alpha_2 s^2 + \alpha_3 s + \alpha_4}$$

Proof

We now show that the pair

$$\left(\begin{bmatrix} A & 0 \\ -c & 0 \end{bmatrix}, \begin{bmatrix} b \\ 0 \end{bmatrix} \right) \quad \text{(New pair)}$$

is controllable if and only if $\beta_4 \neq 0$.

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Its determinant is $-\beta_4$. Thus the matrix is nonsingular if and only if $\beta_4 \neq 0$.

Robust Controller Design (1/5)

Consider again (Aug. Dist. CL-LTI)

$$\begin{bmatrix} \dot{x} \\ \dot{x}_a \end{bmatrix} = \begin{bmatrix} A + bk & bk_a \\ -c & 0 \end{bmatrix} \begin{bmatrix} x \\ x_a \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} r + \begin{bmatrix} b \\ 0 \end{bmatrix} w$$
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Assume that a set of $n + 1$ desired stable eigenvalues has been selected and the feedback gain $[k \quad k_a]$ has been found such that

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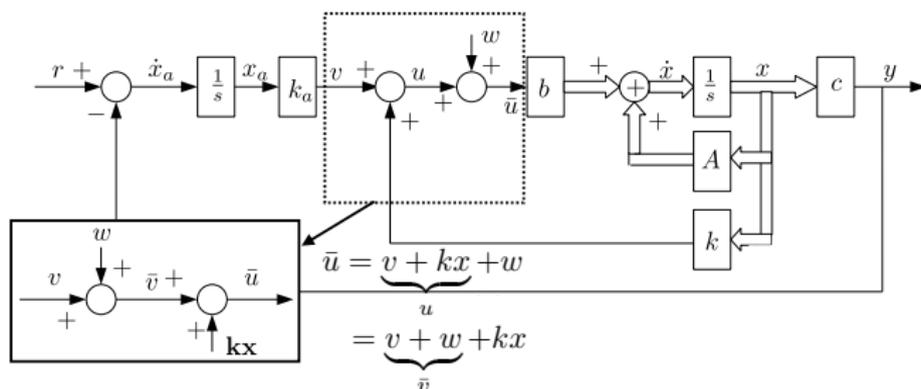
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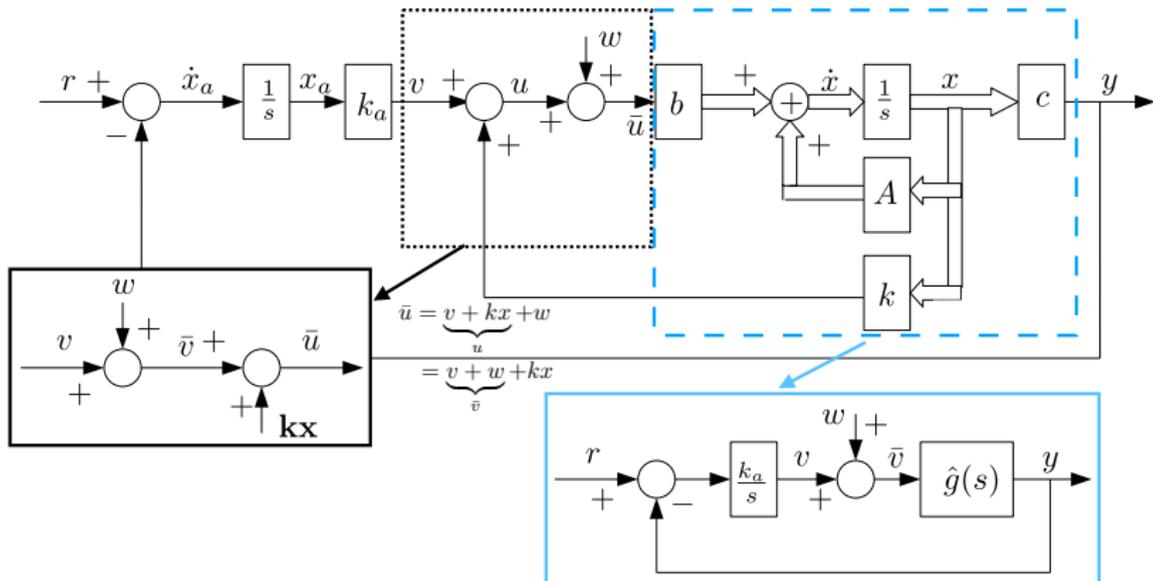


Robust Controller Design (2/5)

The transfer function from \bar{v} to y is

$$\hat{g}(s) := \frac{\bar{N}(s)}{\bar{D}(s)} := c(sI - A - bk)^{-1}b$$

with $\bar{D}(s) = \det(sI - A - bk)$.



Robust Controller Design (3/5)

It is straight forward to verify the following equality:

$$\underbrace{\begin{bmatrix} I & 0 \\ c(sI - A - bk)^{-1} & 1 \end{bmatrix}}_{\text{unimodular}} \overbrace{\begin{bmatrix} sI - A - bk & -bk_a \\ c & s \end{bmatrix}}^{(sI - A_{\text{ADCL-LTI}})} = \begin{bmatrix} sI - A - bk & -bk_a \\ 0 & s + c(sI - A - bk)^{-1}bk_a \end{bmatrix}$$

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Taking its determinant, we obtain

$$1 \cdot \Delta_f(s) = \bar{D}(s) \left(s + \frac{\bar{N}(s)}{\bar{D}(s)} k_a \right) = s\bar{D}(s) + k_a \bar{N}(s).$$

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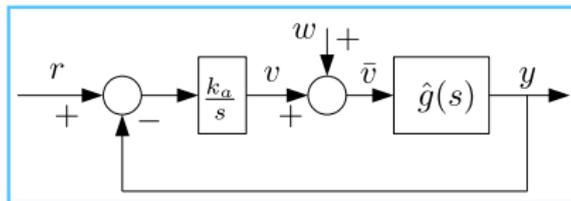
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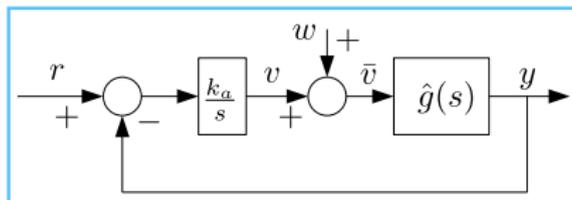
$$\Delta_f(s) = s\bar{D}(s) + k_a \bar{N}(s)$$

This is a key equation.

Robust Controller Design (4/5)

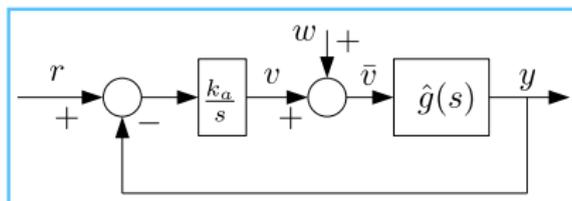


Robust Controller Design (4/5)



$$\hat{g}_{yw} = \frac{\bar{N}(s)}{D(s)} = \frac{s\bar{N}(s)}{s\bar{D}(s) + k_a\bar{N}(s)} = \frac{s\bar{N}}{\Delta_f(s)}$$

Robust Controller Design (4/5)

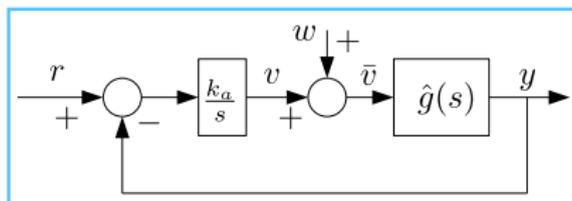


$$\hat{g}_{yw} = \frac{\frac{\bar{N}(s)}{D(s)}}{1 + \frac{k_a}{s} \frac{\bar{N}(s)}{D(s)}} = \frac{s\bar{N}(s)}{s\bar{D}(s) + k_a\bar{N}(s)} = \frac{s\bar{N}}{\Delta_f(s)}$$

If the disturbance is $w(t) = \bar{w}$ for all $t \geq 0$, where \bar{w} is unknown constant, then $\hat{w}(s) = \bar{w}/s$ and the corresponding output is given by

$$\hat{y}_w = \frac{s\bar{N}(s)}{\Delta_f(s)} \frac{\bar{w}}{s} = \frac{\bar{w}\bar{N}(s)}{\Delta_f(s)}$$

Robust Controller Design (4/5)



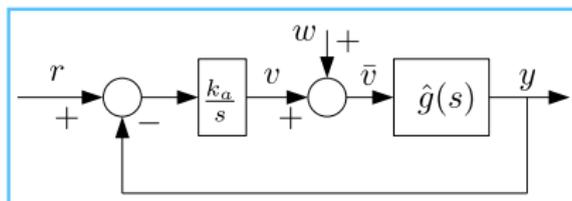
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Because the pole s is cancelled, all remaining poles of $\hat{y}_w(s)$ are stable poles. Therefore the corresponding time response, for any \bar{w} , will die out as $t \rightarrow \infty$.

Robust Controller Design (4/5)



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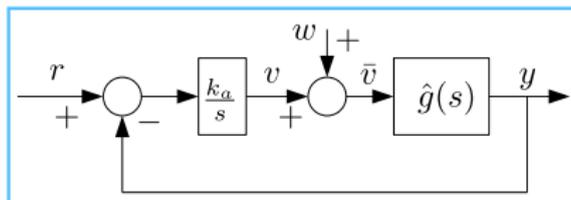
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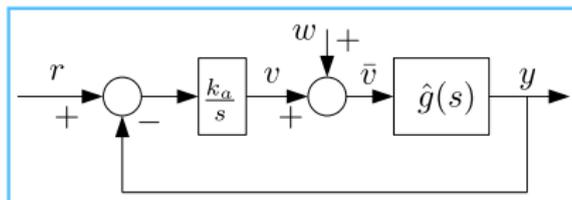
If there are plant parameter variations and variations in the feedforward gain k_a and feedback gain k , the rejection still holds as long as overall system remains stable.

Robust Controller Design (5/5)



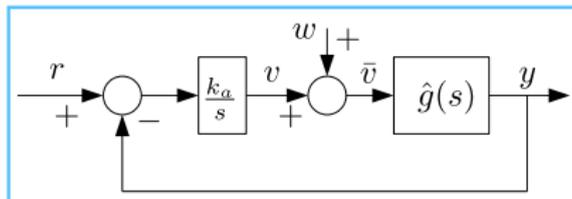
$$\hat{g}_{yr}(s)$$

Robust Controller Design (5/5)



$$\hat{g}_{yr}(s) = \frac{\frac{k_a \bar{N}(s)}{s \bar{D}(s)}}{1 + \frac{k_a \bar{N}(s)}{s \bar{D}(s)}} = \frac{k_a \bar{N}(s)}{s \bar{D}(s) + k_a \bar{N}(s)} = \frac{k_a \bar{N}(s)}{\Delta_f(s)}$$

Robust Controller Design (5/5)

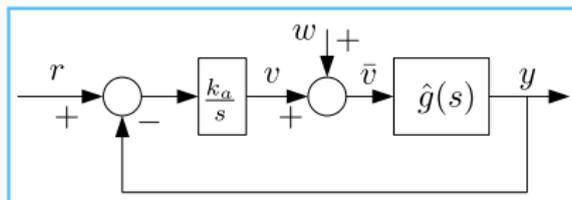


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We see that

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Robust Controller Design (5/5)



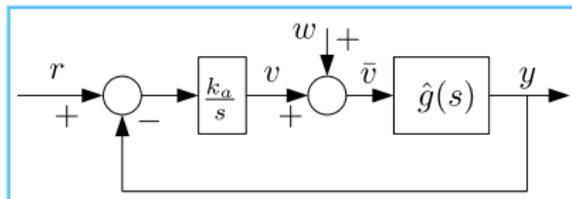
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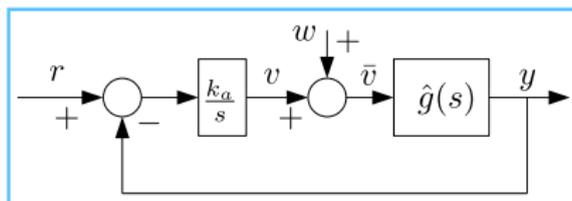
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Note that this robust tracking holds even for very large parameter perturbations as long as overall system remains stable.

Key Observation

The integrator is in fact a model of the step reference input and constant disturbance. Thus it is called the *internal model principle*.

Stabilization

Stabilization problem

If a state equation is controllable, all eigenvalues can be assigned arbitrarily by introducing the state feedback. The problem is to design a *stabilizing controller* whenever the state equation is not state-controllable .

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$$\begin{bmatrix} \dot{\bar{x}}_c \\ \dot{\bar{x}}_u \end{bmatrix} = \begin{bmatrix} \bar{A}_c & \bar{A}_{12} \\ 0 & \bar{A}_u \end{bmatrix} \begin{bmatrix} \bar{x}_c \\ \bar{x}_u \end{bmatrix} + \begin{bmatrix} \bar{b}_c \\ 0 \end{bmatrix} u \quad (\text{Uncontrollable Decomposition})$$

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Observation

Because the “new A -matrix” is block triangular, the eigenvalues of the original A -matrix are the union of the eigenvalues of \bar{A}_c and \bar{A}_u .

Stabilization

Let introduce the state feedback controller

$$u = r - kx = r - \bar{k}\bar{x} = r - \begin{bmatrix} \bar{k}_1 & \bar{k}_2 \end{bmatrix} \begin{bmatrix} \bar{x}_c \\ \bar{x}_u \end{bmatrix}$$

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We see that \bar{A}_u and consequently, its eigenvalues are not affected by the state feedback. Thus we conclude that controllability condition of (A, b) is not only sufficient (as stated earlier, see slide #13) but also necessary to assign all eigenvalues of $(A - bk)$ to any desired positions.

Stabilization

Recall stabilizability

Consider again the (Uncontrollable Decomposition) state equation. If \bar{A}_u is stable, and if (\bar{A}_c, \bar{b}_c) is controllable then the state equation is said to be stabilizable.

Comment on Tracking and Rejection problems

- 1 The controllability condition for tracking and disturbance rejection can be replaced by the *weaker* condition of stabilizability.
- 2 But, we do not have complete control of the *rate of tracking and rejection*.
- 3 If the uncontrollable stable eigenvalues have large imaginary parts or are close to imaginary axis, then the tracking and rejection *may not* be satisfactory.

State feedback

Consider a plant described by the n -dimensional p -input state equation

$$\begin{aligned}\dot{x} &= Ax + Bu \\ y &= Cx\end{aligned}\tag{Plant}$$

In state feedback, the input u is given by

$$u = r - Kx\tag{Controller}$$

where K is a $p \times n$ real constant matrix and r is a reference signal. Substituting (Controller) in (Plant) yields

$$\begin{aligned}\dot{x} &= (A - BK)x + Br \\ y &= Cx\end{aligned}\tag{Closed-loop}$$

State feedback

Theorem

The pair $(A - BK, B)$, for any $p \times n$ real constant matrix K , is controllable if and only if (A, B) is controllable.

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Logical idea.

- The proof of this theorem follows closely the proof of the earlier result. The only difference is that we must modify the key equation as

$$\mathfrak{C}_f = \mathfrak{C} \begin{bmatrix} I_p & -KB & -K(A - BK)B & -K(A - BK)^2B \\ 0 & I_p & -KB & -K(A - BK)B \\ 0 & 0 & I_p & -KB \\ 0 & 0 & 0 & I_p \end{bmatrix}$$

where \mathfrak{C}_f and \mathfrak{C} are $n \times np$ controllability matrices with $n = 4$ and I_p is the unit matrix of order p .

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where \mathfrak{C}_f and \mathfrak{C} are $n \times np$ controllability matrices with $n = 4$ and I_p is the unit matrix of order p .

- 2 Because the rightmost $4p \times 4p$ matrix is nonsingular, \mathfrak{C}_f has rank n if and only if \mathfrak{C} has rank n . Thus the controllability property is preserved in any state feedback.

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*All eigenvalues of $(A - BK)$ can be assigned **arbitrarily** (provided complex conjugate eigenvalues assigned in pairs) by selecting a real constant K if and only if (A, B) is controllable.*

If (A, B) is not controllable, then (A, B) can be transformed into the form shown in (Uncontrollable Decomposition) and the eigenvalues of \bar{A}_u will not be affected by any state feedback.

Lyapunov-Equation Method

Problem

Consider an n -dimensional p -input pair (A, B) . Find a $p \times n$ real constant matrix K so that $(A - BK)$ has any set of desired eigenvalues as long as the set does not contain any eigenvalue of A .

Lyapunov-Equation Method

- 1 Select an $n \times n$ matrix F with a set of desired eigenvalues that contains no eigenvalues of A .
- 2 Select an arbitrary $p \times n$ matrix \bar{K} such that (F', \bar{K}') is controllable.
- 3 Solve the unique T in the Lyapunov equation $AT - TF = B\bar{K}$.

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 - If T is nonsingular, the Lyapunov equation and $KT = \bar{K}$ imply

$$(A - BK)T = TF \text{ or } A - BK = TFT^{-1}$$

Thus $A - BK$ and F are similar and have the same set of eigenvalues.

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$$\det T \neq 0 \begin{array}{c} \xrightarrow{\text{nec}} \\ \xleftarrow{\text{suf}} \end{array} \text{rank} \mathfrak{C}_{(A,B)} = \text{rank} \mathfrak{C}_{(F', \bar{K}')} = n$$

Lyapunov-Equation Method

Theorem

If A and F have no eigenvalues in common, then the unique solution T of $AT - TF = B\bar{K}$ is nonsingular only if the pairs (A, B) and (F', \bar{K}') are controllable.

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Proof.

The proof is similar to that of the previous except that

$$\Delta(A)T - T\Delta(F) = -T\Delta(F) = [b \quad Ab \quad A^2b \quad A^3b] \begin{bmatrix} \alpha_3 & \alpha_2 & \alpha_1 & 1 \\ \alpha_2 & \alpha_1 & 1 & 0 \\ \alpha_1 & 1 & 0 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} \bar{k} \\ \bar{k}F \\ \bar{k}F^2 \\ \bar{k}F^3 \end{bmatrix}$$

now modifies to

$$\begin{aligned} &= [B \quad AB \quad A^2B \quad A^3B] \begin{bmatrix} \alpha_3 I & \alpha_2 I & \alpha_1 I & I \\ \alpha_2 I & \alpha_1 I & I & 0 \\ \alpha_1 I & I & 0 & 0 \\ I & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} \bar{K} \\ \bar{K}F \\ \bar{K}F^2 \\ \bar{K}F^3 \end{bmatrix} \\ &= \mathfrak{C}_{(A,B)} \Sigma \mathfrak{C}_{(F', \bar{K}')} \end{aligned}$$

where $\Delta(F)$ is nonsingular and $\mathfrak{C}_{(A,B)}$, Σ , and $\mathfrak{C}_{(F', \bar{K}')}$ are, respectively, $n \times np$, $np \times np$ and $np \times n$. If $\mathfrak{C}_{(A,B)}$ or $\mathfrak{C}_{(F', \bar{K}')}$ has rank less than n , then T is singular. However the conditions that $\mathfrak{C}_{(A,B)}$ and $\mathfrak{C}_{(F', \bar{K}')}$ have rank n do not imply the nonsingularity of T . Thus the controllability of (A, B) and (F', \bar{K}') are only necessary conditions for T to be nonsingular. \square

Lyapunov-Equation Method

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Definition (Cyclic matrix)

A matrix A is called *cyclic* whenever the Jordan form of A has one and only Jordan block associated with each distinct eigenvalue.

Theorem

If the n -dimensional p -input pair (A, B) is controllable and if A is cyclic, then for almost any $p \times l$ vector v , the single-input pair (A, Bv) is controllable.

Cyclic Design

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Controllability Invariance

Controllability is invariant under any equivalence transformation; thus we may assume A to be in Jordan form.

Cyclic Design: Logical idea behind the proof

To see the basic idea, we use the following example:

$$A = \begin{bmatrix} 2 & 1 & 0 & 0 & 0 \\ 0 & 2 & 1 & 0 & 0 \\ 0 & 0 & 2 & 0 & 0 \\ 0 & 0 & 0 & -1 & 1 \\ 0 & 0 & 0 & 0 & -1 \end{bmatrix} \quad B = \begin{bmatrix} 0 & 1 \\ 0 & 0 \\ 1 & 2 \\ 4 & 3 \\ 1 & 0 \end{bmatrix} \quad Bv = B \begin{bmatrix} v_1 \\ v_2 \end{bmatrix} = \begin{bmatrix} \times \\ \times \\ \alpha \\ \times \\ \beta \end{bmatrix}$$

There is only one Jordan block associated with each distinct eigenvalue; thus A is cyclic.

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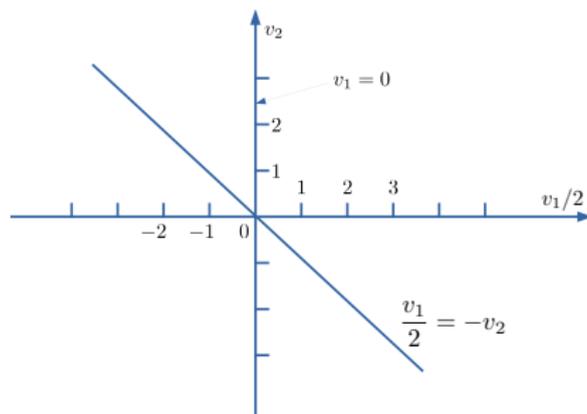
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Because $\alpha = v_1 + 2v_2$ and $\beta = v_1$, either α or β is zero if and only if $v_1 = 0$ or $v_1/v_2 = -2/1$. Thus any v other than $v_1 = 0$ and $v_1 = -2v_2$ will make (A, Bv) controllable.

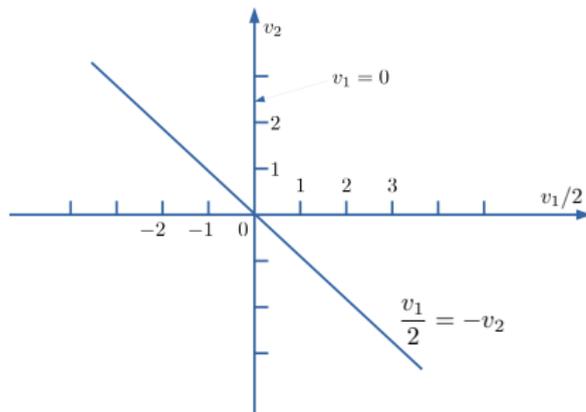
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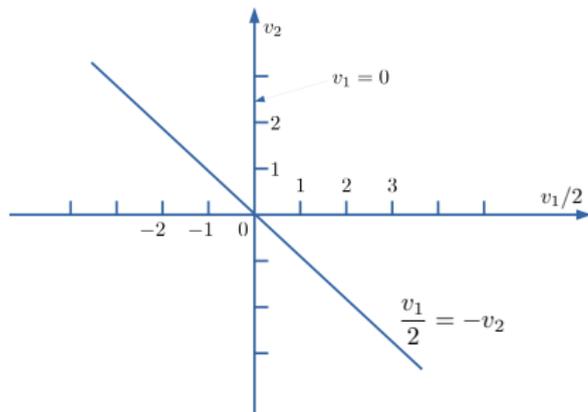
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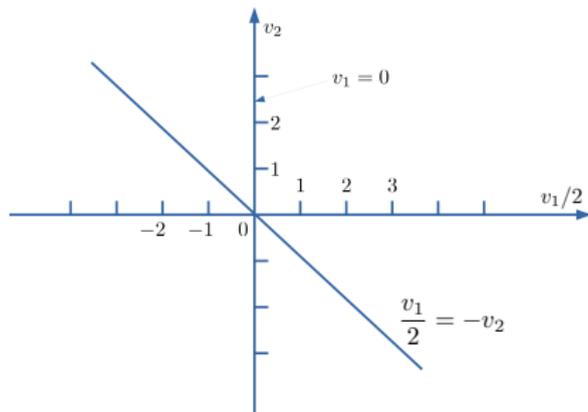
The cyclicity assumption in this theorem is essential. For example, the pair

$$A = \begin{bmatrix} 2 & 1 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 2 \end{bmatrix}, \quad B = \begin{bmatrix} 2 & 1 \\ 0 & 2 \\ 1 & 0 \end{bmatrix}$$

is controllable. However, there is no v such that (A, Bv) is controllable.

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is controllable. However, there is no v such that (A, Bv) is controllable.

If all eigenvalues of A are distinct, then there is only one Jordan block associated with each eigenvalue. Thus a sufficient condition for A to be cyclic is that all eigenvalues of A are distinct.

Theorem

If (A, B) is controllable, then for almost any $p \times n$ real constant matrix K , the matrix $(A - BK)$ has only distinct eigenvalues and is, consequently cyclic.

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If the n -dimensional p -input pair (A, B) is controllable and if A is cyclic, then for almost any $p \times l$ vector v , the single-input pair (A, Bv) is controllable.

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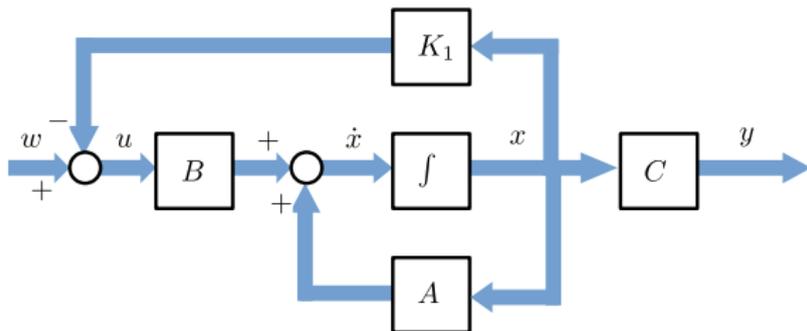
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With these two theorems, we can now find a K to place all eigenvalues of $(A - BK)$ in any desired positions.

If A is not cyclic, we introduce $u = w - K_1x$, such that $\bar{A} := A - BK_1$ in

$$\dot{x} = (A - BK_1)x + Bw =: \bar{A}x + Bw$$

is cyclic.



Cyclic Design

Because (A, B) is controllable, so is (\bar{A}, B) . Thus there exists a $p \times 1$ real vector v such that (\bar{A}, Bv) is controllable¹.

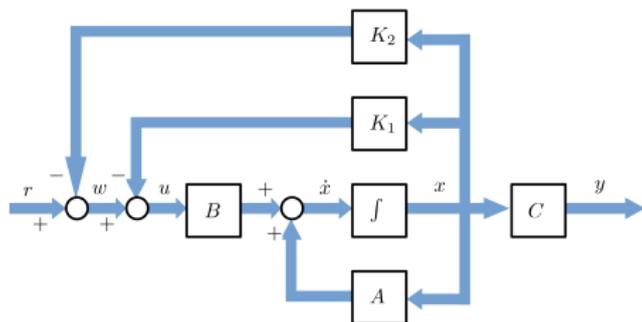
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$$\dot{x} = (\bar{A} - BK_2)x + Br = (\bar{A} - Bvk)x + Br$$



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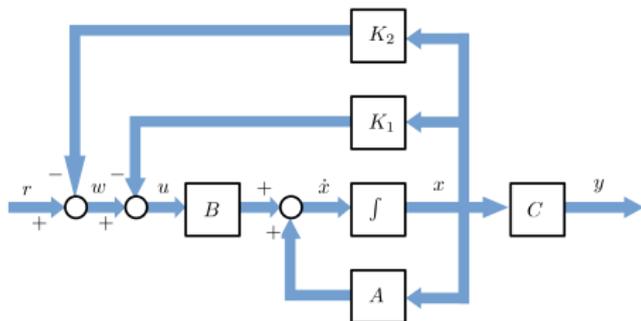
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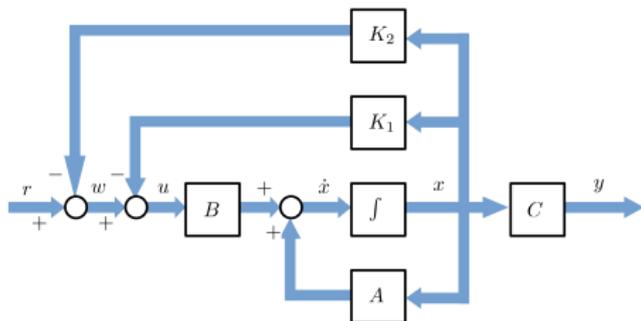
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Combining the two state feedback $u = w - K_1x$ and $w = r - K_2x$ as

$$u = r - (K_1 + K_2)x =: r - Kx$$

we obtain a $K := K_1 + K_2$ that achieves arbitrary eigenvalue assignment.



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The Linear Quadratic Regulator Problem

Problem

Given a continuous-time LTI system:

$$\dot{x} = Ax + Bu, \quad y = Cx$$

the *linear quadratic regulation (LQR)* problem consists of finding the control signal $u(t)$ that makes the following criterion as small as possible:

$$J_{LQR} \triangleq \int_0^{\infty} y^T(t)Qy(t) + u^T(t)Ru(t)dt, \quad (\text{Cost function})$$

where Q and R are the positive-definite weighting matrices.

The following terms provides a *measure*

$$\int_0^{\infty} y^T(t)Qy(t) \quad (\text{Output energy})$$

$$\int_0^{\infty} u^T(t)Ru(t) \quad (\text{Control energy})$$

The Linear Quadratic Regulator Problem

$$J_{LQR} \triangleq \int_0^{\infty} y^T(t)Qy(t) + u^T(t)Ru(t)dt$$

In LQR one seeks a controller that minimizes both energies. However, decreasing the energy of the output requires a large control signal, and a small control signal leads to large outputs.

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In LQR one seeks a controller that minimizes both energies. However, decreasing the energy of the output requires a large control signal, and a small control signal leads to large outputs.

The role of the weighting matrices Q and R is to establish a trade-off between these two conflicting goals.

- 1 When R is much larger than Q , the most effective way to decrease J_{LQR} is to employ a small control input at the expense of a large output.
- 2 When R is much smaller than Q , the most effective way to decrease J_{LQR} is to obtain a very small output, even if this is achieved at the expense of employing a large control input.

Feedback Invariants

Definition (Feedback Invariant)

Given a continuous-time LTI system:

$$\dot{x} = Ax + Bu, \quad y = Cx, \quad x \in \mathbb{R}^n, u \in \mathbb{R}^k, y \in \mathbb{R}^m \quad (\text{CLTI})$$

we say that a functional

$$H(x(\cdot), u(\cdot))$$

that involves the system's input and state is a *feedback invariant* for the system (CLTI) whenever its value depends only on the initial condition $x(0)$ and not on the specific input $u(\cdot)$.

Feedback Invariants

Theorem (Feedback Invariant)

For a symmetric matrix P , the functional

$$H(x(\cdot), u(\cdot)) \triangleq - \int_0^{\infty} (Ax(t) + Bu(t))^T P x(t) + x^T(t) P (Ax(t) + Bu(t)) dt$$

is a feedback invariant for CLTI as long as $\lim_{t \rightarrow \infty} x(t) = 0$.

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Proof.

We can write H as

$$\begin{aligned} H(x(\cdot), u(\cdot)) &= - \int_0^{\infty} \dot{x}^T(t) P x(t) + x^T(t) P \dot{x}(t) dt \\ &= - \int_0^{\infty} \frac{d(x^T P x)}{dt} \\ &= x(0)^T P x(0) - \lim_{t \rightarrow \infty} x^T P x = x^T(0) P x(0), \end{aligned}$$

as long as $\lim_{t \rightarrow \infty} x(t) = 0$. □

Feedback Invariants in Optimal Control

Suppose that we are able to express a criterion J to be minimized by an appropriate choice of the input $u(\cdot)$ in the following form:

$$J = H(x(\cdot), u(\cdot)) + \int_0^{\infty} \Lambda(x(t), u(t)) dt, \quad (\text{criterion})$$

where H is a feedback invariant and the function $\Lambda(x, u)$ has the property that for every $x \in \mathbb{R}^n$

$$\min_{u \in \mathbb{R}^k} \Lambda(x, u) = 0$$

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Note that it is not possible to get a lower value for J since

- ① the feedback invariant H is never affected by u and
- ② a smaller value for J would require the integral in the right hand side of (criterion) to be negative, which is not possible since $\Lambda(x(t), u(t))$ can at best be as low as zero.

Optimal State Feedback

$$J = H(x(\cdot), u(\cdot)) + \int_0^{\infty} \Lambda(x(t), u(t)) dt, \quad (\text{criterion})$$

It turns out that the LQR criterion can be expressed as in (criterion) for an appropriate choice of feedback invariant.

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$$\begin{aligned} J_{LQR} &= \int_0^{\infty} \left(x^T C^T Q C^T x + u^T R u \right) dt \\ &= H + \int_0^{\infty} \left(x^T C^T Q C^T x + u^T R u + (Ax + Bu)^T P x + x^T P (Ax + Bu) \right) dt \end{aligned}$$

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By completing the squares as follows, we group the quadratic term in u with the cross-term in u times x

$$\begin{aligned} \left(u^T + x^T K^T \right) R (u + Kx) &= u^T R u + x^T P B R^{-1} B^T P x + 2u^T B^T P x, \\ K &= R^{-1} B^T P. \end{aligned}$$

Optimal State Feedback

We conclude that

$$J_{LQR} = H(x(\cdot), u(\cdot)) + \int_0^{\infty} x^T \left(A^T P + PA + C^T Q C^T - P B R^{-1} B^T P \right) x \\ + \left(u^T + x^T K^T \right) R (u + Kx) dt.$$

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If we are able to select the matrix P so that

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leading to the closed-loop system

$$\dot{x} = A x + B K x = \left(A - B R^{-1} B^T P \right) x.$$

Optimal state feedback

Theorem (Optimal State Feedback)

Assume that there exists a symmetric matrix P to the following algebraic Riccati equation (ARE)

$$A^T P + PA + C^T Q C - P B R^{-1} B^T P = 0 \quad (\text{ARE})$$

for which $A - B R^{-1} B^T P$ is a stability matrix. Then the feedback control law

$$u = -Kx, \quad K = R^{-1} B^T P,$$

stabilizes the closed-loop system while minimizing the LQR criterion

$$J_{LQR} \triangleq \int_0^{\infty} y^T(t) Q y(t) + u^T(t) R u(t) dt.$$

Note: Asymptotic stability of the closed loop system is needed because we assumed that $\lim_{t \rightarrow \infty} x(t) P x(t) = 0$.

¹Kumar and Jain. Some Insights on Synthesizing Optimal Linear Quadratic Controllers using Krotov Sufficient Conditions, IEEE Control Systems Letters, 2020.

Optimal state feedback

Attention

The ARE itself already provides the clues about whether or not the closed-loop system is stable. Indeed if we write the Lyapunov equation for the closed loop, we get

$$\begin{aligned} (A - BR^{-1}B^T P)^T P + P(A - BR^{-1}B^T P) \\ = A^T P + PA - 2PBR^{-1}B^T P \\ = -\bar{Q} \leq 0 \end{aligned}$$

for $\bar{Q} = C^T Q C + PBR^{-1}B^T P \geq 0$. In case $P > 0$ and $\bar{Q} > 0$, we could immediately conclude that the closed loop system was stable by Lyapunov stability theorem.

LQR with MATLAB

The command

$$[K,P,E] = \text{lqr}(A,B,QQ,RR,NN)$$

computes the optimal state feedback LQR controller for the process

$$\dot{x} = Ax + Bu$$

with the criterion

$$J = \int_0^{\infty} x(t)'QQx(t) + u(t)'RRu(t) + 2x(t)'NNu(t)dt.$$

For the criterion in (Cost function), one should select

$$QQ = C'QC, \quad RR = R, \quad NN = 0.$$

This command returns the optimal state feedback matrix K , the solution P to the corresponding algebraic Riccati equation, and the poles E of the closed-loop system.