

## Unit 2 - Week 0

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## Assignment 0

The due date for submitting this assignment has passed.  
As per our records you have not submitted this assignment.

**Due on 2020-09-14, 23:59 IST.**

This Assignment 0 is a Non Graded Assignment, just to gauge the depth of understanding the learners are having on the pre-requisites for the course.

The following questions may have more than one correct answers. Read and analyse the question carefully before selecting the answer (s).

1) **Wave Drag can be reduced by using :** 1 point

- Thin Wings
- Wing Sweep
- Supercritical airfoils
- Winglets

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*Thin Wings*  
*Wing Sweep*  
*Supercritical airfoils*

2) **Which of the following statements is/are TRUE as the altitude increases in the Troposphere of International Standard Atmosphere?** 1 point

- Temperature decreases, and Dynamic Viscosity decreases
- Temperature remains constant, and Pressure increases
- Temperature decreases, and Sonic Speed decreases
- Temperature increases, and Dynamic Viscosity increases

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*Temperature decreases, and Dynamic Viscosity decreases*  
*Temperature decreases, and Sonic Speed decreases*

3) **Which of the following statement(s) is/are TRUE about the Auxiliary Power Unit?** 1 point

The Auxiliary Power Unit (APU) is utilized to provide power for

- Operating the aircraft from short runways
- Starting up the engine for takeoff
- Air conditioning of the passenger cabin when engine is switched off after landing
- Air conditioning of the passenger cabin when engine is switched off before takeoff

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*Starting up the engine for takeoff*  
*Air conditioning of the passenger cabin when engine is switched off after landing*  
*Air conditioning of the passenger cabin when engine is switched off before takeoff*

4) **Which of the following is favorable for an airplane operation?** 1 point

- Tailwind during Cruise, but Head wind during Landing
- Tailwind during both Cruise and Landing
- Headwind during both Cruise and Landing
- Headwind during Cruise, but Tail wind during Landing

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*Tailwind during Cruise, but Head wind during Landing*

5) **Thrust available from a Turbojet engine** 1 point

- Increases as altitude increases
- Increases up to the tropopause and then decreases
- Remains constant at all altitudes
- Decreases as altitude increases

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*Decreases as altitude increases*

6) **The Pitot tube of an aircraft registers a total pressure  $P_0 = 54,051 \text{ N/m}^2$ . The static pressure =  $45,565 \text{ N/m}^2$ , density =  $0.6417 \text{ kg/m}^3$  and the ratio of specific heats of the freestream is 1.4. The indicated airspeed (in m/s) is** 1 point

- 157.6 m/s
- 162.6 m/s
- 172.0 m/s
- 182.3 m/s

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*162.6 m/s*

7) **The absolute ceiling of an aircraft is the altitude above which it** 1 point

- Can never reach
- Cannot sustain level flight at a constant speed
- Can perform accelerated flight as well as straight and level flight at constant speed
- Can perform straight and level flight at a constant speed only

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*Cannot sustain level flight at a constant speed*

8) **The minimum velocity at the steady level flight of the aircraft corresponding to the load factor  $n = 1$  is termed as** 1 point

- Cruise Speed
- Takeoff Safety Speed
- Unstick Speed
- Stall Speed

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*Stall Speed*

9) **The Drag Divergence Mach ( $M_{DD}$ ) number of an airfoil is** 1 point

- a fixed number for a given airfoil
- always higher than the Critical Mach number ( $M_{crit}$ )
- equal to the critical Mach number ( $M_{crit}$ ) at zero angle of attack
- the Mach number at which a shock wave first appears on the airfoil

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*always higher than the Critical Mach number ( $M_{crit}$ )*

10) **Which of the following statements is/are TRUE with respect to Supercritical Aerofoils?** 1 point

- They have a large Leading edge radius
- They have a flat upper surface
- They have a reflex camber at trailing edge
- They delay the onset of wave drag

No, the answer is incorrect.  
Score: 0

Accepted Answers:

*They have a large Leading edge radius*  
*They have a flat upper surface*  
*They have a reflex camber at trailing edge*  
*They delay the onset of wave drag*