

## National Program on Technology Enhanced Learning

### CONDUCTION AND RADIATION

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#### Final Examination

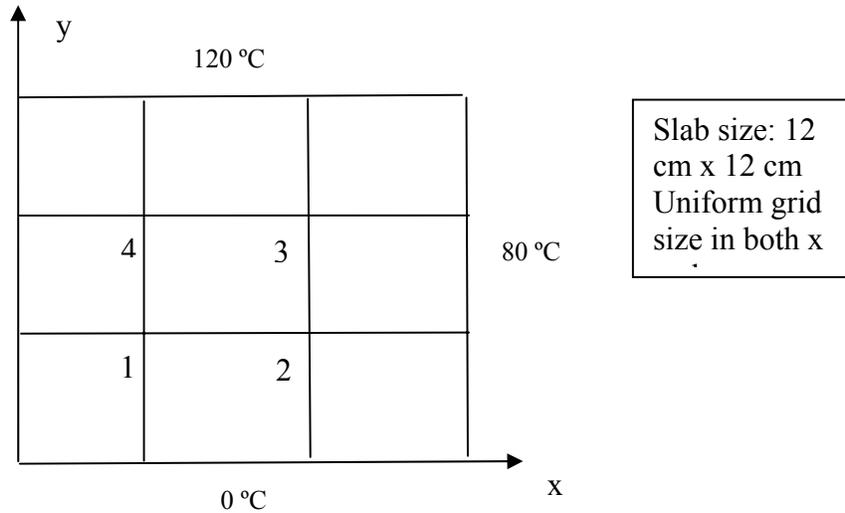
Max. Marks: 100

Duration: 3 hours

1. Make suitable assumptions wherever required with justification
  2. Assume any missing data
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- (1) The steady state temperature distribution in a one dimensional wall of constant thermal conductivity of  $50 \text{ W/mK}$  and thickness  $45 \text{ mm}$  is observed to be  $T = 200 - 2100 x^2$  where  $T$  is in  $^\circ \text{C}$  and  $x$  is in metres (measured from the left end of the wall).
  - (a) What is the volumetric heat generation,  $q_v$  in  $\text{W/m}^3$ .
  - (b) Determine the heat flux at the two walls.
  - (c) What is the relationship between the wall heat fluxes and the heat generation rate? (6)
- (2) A Bakelite coating is to be used with a  $10 \text{ mm}$  diameter conducting rod, whose surface is maintained at  $240^\circ \text{C}$  by passage of an electrical current. The rod is in contact with a fluid at  $30^\circ \text{C}$  and the convection heat transfer coefficient is  $150 \text{ W/m}^2\text{K}$ . How much Bakelite should be added to reduce the heat transfer rate associated with the bare rod by  $30\%$  (thermal conductivity of Bakelite is  $1.4 \text{ W/mK}$ ) (8)
- (3) A composite spherical shell of inner radius  $r_1 = 0.25 \text{ m}$  is constructed from lead of outer radius  $r_2 = 0.30 \text{ m}$  and stainless steel of outer radius  $r_3 = 0.31 \text{ m}$ . The cavity is filled with radioactive wastes that generate heat at the rate of  $q_v = 5 \times 10^5 \text{ W/m}^3$ . It is proposed to submerge the container in sea waters at a temperature of  $T_\infty = 15^\circ \text{C}$ . The heat transfer coefficient provided by the sea water is  $550 \text{ W/m}^2\text{K}$  at the outer surface of the container. Given that the melting point of lead is  $600 \text{ K}$ , determine if the proposal is safe ( $k_{\text{pb}} = 35 \text{ W/mK}$ ,  $k_{\text{SS}} = 15.1 \text{ W/mK}$ ). (8)
- (4) A thin metallic wire of thermal conductivity  $k$ , diameter  $D$  and length  $2L$  is annealed by passing an electric current through the wire in order to induce a uniform volumetric heat generation  $q_v$ . The heat transfer coefficient between the ambient and the wire is given by  $h$ . The ambient temperature and the temperature at the two ends of the wire (i.e. at  $x = \pm L$ ) is  $T_\infty$ .

- (a) Derive the governing equation for the one dimensional temperature distribution in the wire ( $T=f(x)$  only) treating it as a fin that also has heat generation for steady state conditions in the fin.
- (b) Obtain an expression for the steady state temperature distribution in the wire by solving the equation derived in part (a) (6+6)



- (5) Two dimensional, steady state conduction in a square slab with  $k = 40 \text{ W/mK}$  and an internal heat generation of  $q_v = 6 \times 10^5 \text{ W/m}^3$  is to be numerically simulated. The details are given in the figure. For simplicity the number of grid points is intentionally kept small for this problem.
- (a) Identify the governing equation for the problem. (1)
- (b) Using the central difference finite difference scheme in conjunction with the Gauss-Seidel method, estimate  $T_1, T_2, T_3$  and  $T_4$ . Start with an initial guess of  $50^\circ\text{C}$  for all the four temperatures. **Do at least 6 iterations.** (8)
- (c) What is the center temperature? (1)
- (6) Radiation is the only mode of heat transfer possible in space radiators used in space stations. As the sizes of the space stations keep increasing, cooling technologies have also to keep pace. One novel idea is to use what is now known as a Liquid Droplet Radiator (LDR). Here, the heat is first transferred to a high vacuum oil, which is then injected into outer space as a stream of small droplets. The stream travels over a distance “L” over which it cools by radiating heat to

outer space at a temperature  $T_\infty$ . The droplets are returned and collected to the space station.

- (a) Derive the governing equation for the transient temperature response of a spherical droplet of mass  $m$ , specific heat capacity  $C_p$  and emissivity  $\epsilon$ . The droplet is at an initial temperature of  $T_i$ . You may assume the droplet to be spatially isothermal.
- (b) Solve the governing equation and obtain an expression for the transient temperature response of the droplet.
- (c) Consider droplets that have an emissivity of 0.95 and diameter of 0.6 mm. These are injected at an initial temperature of  $T_i = 520$  K and a velocity of 0.11 m/s. Properties of the oil are  $\rho=890$  kg/m<sup>3</sup>,  $C_p= 1900$  J/kgK and  $k=0.15$  W/mK. Assume each droplet to radiate to deep space at  $T_\infty = 0$  K. Determine the distance  $L$  for the droplets to reach the collector at a final temperature of 303 K. What is the enthalpy rejected by each droplet in the process?
- (d) Is the lumped capacitance formulation justified? (4+6+7+1)
- (7) Consider a thin opaque, horizontal plate with an electrical heater on its bottom side. The top side is exposed to ambient air at 25 ° C with a convection heat transfer coefficient of 12 W/m<sup>2</sup>K, solar irradiation of 650 W/m<sup>2</sup> and an effective sky temperature of -40 ° C. Determine the electrical power required to maintain the temperature of the surface at 65 ° C if the plate is diffuse and has the following spectral, hemispherical reflectivity

$$0 \leq \lambda \leq 2 \mu\text{m}, \rho_\lambda = 0.2$$

$$2 \leq \lambda \leq \infty \mu\text{m}, \rho_\lambda = 0.75$$

(10)

- (8) A two dimensional evacuated, gray diffuse enclosure is in the shape of a triangle ABC. The sides of the enclosure are infinitely deep in the direction perpendicular to the plane of the paper.

Surface	Length, m	Emissivity, $\epsilon$	Temperature, K
AB	8	0.65	1050
BC	6	0.45	450
CA	9	0.75	800

Compute the net radiation heat transfer from the three surfaces using the radiosity-irradiation method. (14)

- (9) A gas turbine combustion chamber may be approximated as a long tube 40 cm in diameter. The combustion gases (**a participating medium**) are at a pressure of 2 atm and  $1150^{\circ}\text{C}$ , while the surface temperature of the chamber is  $550^{\circ}\text{C}$ . If the combustion gases are composed of  $\text{CO}_2$  with a mole fraction of 0.2, water vapour with a mole fraction of 0.15 and the remainder Nitrogen, what is the net radiation heat transfer between the gas and the surface of the combustion chamber? The emissivity of the combustion chamber surface may be taken to be 0.90. (14)