#### ME-662 CONVECTIVE HEAT AND MASS TRANSFER

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LECTURE-36 EVALUATION OF g and  $N_w$ 

### **LECTURE-36 EVALUATION OF g and** N<sub>w</sub>

- Laminar Boundary Layers
- 2 Turbulent Boundary Layers
- Overall Procedure for calculating N<sub>w</sub>

## Laminar BL - 1 - L36( $\frac{1}{10}$ )

• Consider Laminar BL with  $T_w = \text{const.}$  and with suction/blowing and without viscous dissipation. For this case, Similarity soln for const properties is

$$\frac{Nu_x}{Re_x^{0.5}} = -\theta'(0) = F(m, Pr, B_f)$$

$$Nu_x = \frac{h_x x}{k} = \frac{x (\partial T/\partial y)_w}{T_\infty - T_w} \text{ and } B_f = \frac{V_w}{U_\infty} Re_x^{0.5}$$

2 This corresponds to  $\Psi = T$  and  $\omega_k = 1$  with constant specific heat in all states. Hence, in terms of mass transfer coeff (g)

$$g = \frac{\Gamma_{\Psi} \left( \frac{\partial \Psi}{\partial y} \right)_{W}}{\Psi_{\infty} - \Psi_{W}} \text{ or } Sh_{x} = \frac{g_{x} x}{\Gamma_{\Psi}} = \frac{x \left( \frac{\partial \Psi}{\partial y} \right)_{W}}{\Psi_{\infty} - \Psi_{W}} = Nu_{x}$$

# Laminar BL - 2 - L36( $\frac{2}{10}$ )

Similarly, B<sub>f</sub> can be interpreted as

$$B_{f} = \frac{V_{w}}{U_{\infty}} Re_{x}^{0.5} = \frac{N_{w}}{\rho U_{\infty}} Re_{x}^{0.5} = \frac{g B_{\Psi}}{\rho U_{\infty}} Re_{x}^{0.5}$$
$$= (\frac{g_{x} x}{\Gamma_{\Psi}}) \times (\frac{\Gamma_{\Psi}}{\mu}) \times (\frac{\mu}{\rho U_{\infty} x}) \times Re_{x}^{0.5} B_{\Psi}$$
$$= Sh_{x} (\frac{\Gamma_{\Psi}}{\mu}) Re_{x}^{-0.5} B_{\Psi}$$

2 This shows that driving force  $B_{\Psi} \propto B_f$ . Hence, similarity soln to the  $\Psi$ -eqn can be interpreted as

$$rac{{\sf Sh}_{\sf x}}{{\sf Re}_{\sf x}^{0.5}}={\sf F}({\it m},rac{\mu}{\Gamma_{\Psi}},{\it B}_{\Psi})$$

# Laminar BL - 3 - L36( $\frac{3}{10}$ )

- Using the last relation, the constant property heat transfer solutions (lecture 9 - slide 10) can be converted to mass transfer solutions.
- 2 Thus, consider case of  $B_f = -2$ , m = 0 and Pr =  $\mu/\Gamma_h = 1.0$ .
- For this case,  $Nu_x Re_x^{-0.5} = -\theta'(0) = 2.1 = Sh_x Re_x^{-0.5}$ .
- Hence,

$$B_{\Psi} = rac{(\mu/\Gamma_{\Psi})}{- heta'(0)} imes B_{f} = rac{1}{2.1} imes (-2.0) = -0.9524$$

Solution Next slide shows conversions for  $\mu/\Gamma_{\Psi} = 0.7$  and m = 0

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#### **Laminar BL - 4 - L36(** $\frac{4}{10}$ **)** Conversions for $Sc = \mu/\Gamma_{\Psi} = 0.7 - m = 0$

	m=0				
B <sub>f</sub>	$- heta^{\prime}$ (0)	$B_{\Psi} = rac{\operatorname{Sc}B_{\mathrm{f}}}{- heta^{\prime}\left(0 ight)}$	$rac{g}{g^{*}}=rac{- heta^{\prime}\left(0 ight)}{0.291}$	$\frac{\ln(1+B_{\Psi})}{B_{\Psi}}$	
-2.0	1.52	-0.921	5.223	2.756	
-1.0	0.872	-0.8027	3.00	2.022	
-0.5	0.570	-0.614	1.959	1.55	
-0.25	0.429	-0.4079	1.474	1.285	
0.0	0.291	0.0	1.0	1.0	
0.25	0.166	1.054	0.57	0.683	
0.375	0.107	2.453	0.368	0.505	
0.5	0.0517	6.77	0.1776	0.303	

For  $-0.25 < B_{\Psi} < 0.25$ ,  $(g/g^*) \simeq \ln(1 + B_{\Psi})/B_{\Psi}$ . But, for large  $|B_{\Psi}|$ , the Reynolds flux model is not at all satisfactory. For these cases, numerical solutions are desirable. These observations also apply to other values of m and Sc.

## Laminar BL - 5 - L36( $\frac{5}{10}$ )

For Arbitrarily varying  $U_{\infty}$ , Integral solns (Spalding D B and Chi S W, IJHMT, vol 6, p 363-385 (1963), show that

Stanton<sub>mass</sub> = 
$$\frac{g}{\rho U_{\infty}} = \frac{K_1 \mu^{1.2} (\rho U_{\infty})^{K_2}}{\left[\int_0^x (\rho U_{\infty})^{K_3} dx\right]^{0.5}}$$

Sc	$B_{\Psi}$	<i>K</i> <sub>1</sub>	<b>K</b> <sub>2</sub>	<i>K</i> <sub>3</sub>
	-0.9	1.85	0.05	1.1
0.7	0.0	0.418	0.435	1.87
	9.0	0.06	1.90	4.8
	-0.9	0.431	0.45	1.9
5.0	0.0	0.117	0.595	2.19
	9.0	0.023	0.90	2.8
	-0.9	1.037 Sc <sup>-0.67</sup>	0.9	2.8
> 5	0.0	0.339 Sc <sup>-0.67</sup>	0.9	2.8
	9.0	$0.077 \ Sc^{-0.67}$	0.9	2.8

# Turbulent BL - L36( $\frac{6}{10}$ )

- In Turbulent BLs, the analogy between heat and mass transfer is more perfect because  $\Gamma_{eff} = \Gamma_l + \Gamma_t \simeq \mu_l + \mu_t$  with  $\Gamma_t >> \Gamma_l$  and  $\mu_t >> \mu_l$ . That is, for gases  $\Pr \simeq Sc$  and  $\Pr_t = Sc_t \simeq 0.9$
- 2 The turbulent heat transfer correlations for  $V_w = 0$  take the form of  $St_{x,V_w=0} = C \operatorname{Re}_x^{-m} \operatorname{Pr}^{-n}$ . Then, from analogy,

$$\begin{array}{lll} St_{x,V_{w}=0} & = & \displaystyle \frac{g^{*}}{\rho \; U_{\infty}} = C \; Re_{x}^{-m} \; Sc^{-n} \\ \\ \displaystyle \frac{g}{\rho \; U_{\infty}} & = & \displaystyle \frac{g^{*}}{\rho \; U_{\infty}} \times \displaystyle \frac{\ln(1+B_{\Psi})}{B_{\Psi}} \\ \\ \displaystyle g^{*} & = & \displaystyle \frac{h_{cof,V_{w}=0}}{c_{om}} \; (\frac{Pr}{Sc})^{-n} \quad \rightarrow \quad Sc = \displaystyle \frac{\mu}{\Gamma_{\Psi}} \end{array}$$

# Effect of Property Variations - L36( $\frac{7}{10}$ )

- Deviations from  $(g / g^*) = \ln (1 + B) / B$  at large  $B_{\Psi}$  mainly occur due to property variations through the boundary layer
- Por Laminar BLs, the recommended property-correction is

$$rac{g}{g^*}=rac{\ln(1+B_\Psi)}{B_\Psi} imes(rac{M_w}{M_\infty})^{0.66}$$

So For Turbulent BLs, the recommended property-correction is

$$rac{g}{g^*} = rac{\ln(1+B_\Psi)}{B_\Psi} imes (rac{M_w}{M_\infty})^{0.40}$$

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where M is molecular weight of the mixture. These relations also apply to internal flows where  $B_{\Psi} = (\Psi_b - \Psi_w)/(\Psi_w - \Psi_T)$  and  $\Psi_b$ is the bulk value.

#### **Binary Diffusion Coeffs L36**( $\frac{8}{10}$ ) Binary diffusion coefficient $D_{ab}$ ( $m^2/s$ ) at 1 atm and T = 300 K.

Pair	$D_{ab} imes 10^{6}$	Pair	$D_{ab} imes 10^{6}$
H <sub>2</sub> O-air	24.0	CO <sub>2</sub> -air	14.0
CO-air	19.0	$CO_2 - N_2$	11.0
$H_2$ -air	78.0	O <sub>2</sub> -air	19.0
SO <sub>2</sub> -air	13.0	<i>NH</i> ₃-air	28.0
<i>CH</i> ₃ <i>OH</i> -air	14.0	$C_2H_5OH$ -air	11.0
$C_6H_6$ -air	8.0	<i>CH</i> ₄-air	16.0
$C_{10}H_{22}$ -air	6.0	$C_{10}H_{22}-N_2$	6.4
$C_8H_{18}$ -air	5.0	$C_8H_{18}-N_2$	7.0
$C_8H_{16}-N_2$	7.1	$C_6H_{14}-N_2$	8.0
$O_2$ - $H_2$	70.0	$CO_2$ - $H_2$	55.0

Assuming ideal gas behavior, the kinetic theory of gases predicts that  $D_{ab} \propto (T^{1.5}/p)$ , where T is in Kelvin.

## **Overall Procedure for** $N_w$ L36( $\frac{9}{10}$ )

- For the type of mass transfer problem, identify the appropriate conserved property  $\Psi$ .
- 2 Make sure that  $B_{\Psi}$  can be evaluated from  $\Psi_{\infty}$ ,  $\Psi_{T}$  ( usually known ) and  $\Psi_{w}$  ( usually not known ). If not, select linear combinations of  $\Psi_{S}$ .
- Sometimes,  $\Psi_w$  needs to be established from iterations.
- Identify the heat transfer situation with V<sub>w</sub> = 0 corresponding to the mass transfer problem at hand . Hence, evaluate  $h_{cof, V_w=0}$  and  $g_h^* = h_{cof, V_w=0}/c_{pm}$ .
- Hence, evaluate

$$N_w = g imes B = g_h^* imes (rac{Pr}{Sc})^n imes (rac{M_w}{M_\infty})^x imes \ln{(1+B_\Psi)}$$

where n and x correspond to the problem at hand.

# Summary L36(<sup>10</sup>/<sub>10</sub>)

We have thus examined the validity of

$$egin{array}{rcl} \mathcal{N}_w &=& g imes B_\Psi = g^* \ln \left( 1 + B_\Psi 
ight) ext{ and } \ rac{g}{g^*} &=& \mathcal{F}(\mathcal{B}) = rac{\ln \left( 1 + B_\Psi 
ight)}{B_\Psi} \end{array}$$

2 It is shown that deviations from this formulas occur when fluid properties vary significantly in the boundary layer at large  $B_{\Psi}$ . Hence, the calculation of  $N_w$  is corrected as

$$N_w = g imes B = g_h^* imes (rac{Pr}{Sc})^n imes (rac{M_w}{M_\infty})^x imes \ln{(1+B_\Psi)}$$

where  $g_h^* = h_{cof,V_w=0}/c_{pm}.$ 

In the next 3 lectures, we will demonstrate applications of Stefan-, Couette- and Reynolds-flow models to problems of engineering relevance.

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